Airborne Equipment, are recognized by the FAA as acceptable methodologies for finding compliance with the environmental requirements. Equivalent environment test standards may also be acceptable. Environmental qualification provides data to show that the AP/SAS system can perform its intended function under the expected operating condition. Some of the main considerations for environmental concerns are installation locations and the resulting exposure to environmental conditions for the AP/SAS system equipment, including considerations for other equipment that may also be affected environmentally by the AP/SAS equipment installation. The level of environmental qualification must be related to the severity of the considered failure conditions and effects on the rotorcraft.

Discussion of Comments

Notice of proposed special conditions No. 27-046-SC for the Robinson Model R66 helicopter was published in the Federal Register on June 26, 2019 (84 FR 30050). Comments were received from two commenters. The commenters stated that special conditions previously issued for an AP/SAS system on a different model helicopter were more aligned with rulemaking whereas this special condition text seemed more appropriate for guidance material. The commenters requested the FAA change the proposed special conditions to be consistent with those previously issued for the same type of equipment. The FAA agrees that an effort should be made to maintain consistency and has revised the proposed special conditions to align with previously issued special conditions.

The commenters also noted the Discussion section of the proposed special conditions contains references to specific revisions of RTCA Document DO-178 and RTCA Document DO-160G and requested these references to specific revisions be removed. The FAA disagrees, however an applicant may request to use a later approved revision to these documents if the applicant shows the later revision meets the safety level intended by the special condition.

Except for the changes previously discussed, these special conditions are adopted as proposed.

Applicability

These special conditions are applicable to the AP/SAS installed as an amended TC approval in Robinson Model R66 helicopter, TC Number R00015LA.

Conclusion

This action affects only certain novel or unusual design features for an AP/ SAS amended TC installed on one model helicopter. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features.

List of Subjects in 14 CFR Part 27

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

Accordingly, the Federal Aviation Administration (FAA) issues the following special conditions as part of the amended type certification basis for installation of the autopilot and stability augmentation system (AP/SAS) on Robinson Model R66 helicopters.

Instead of the requirements of 14 CFR 27.1309(b) and (c), the following must be met for certification of the AP/SAS system installed on Robinson Model R66 helicopters:

a. The equipment and systems must be designed and installed so that any equipment and systems do not adversely affect the safety of the rotorcraft or its occupants.

b. The rotorcraft systems and associated components considered separately and in relation to others systems, must be designed and installed so that:

(1) The occurrence of any catastrophic failure condition is extremely improbable;

(2) The occurrence of any hazardous failure condition is extremely remote; and

(3) The occurrence of any major failure condition is remote.

c. Information concerning an unsafe system operating condition must be provided in a timely manner to the crew to enable them to take appropriate corrective action. An appropriate alert must be provided if immediate pilot awareness and immediate or subsequent corrective action is required. Systems and controls, including indications and annunciations, must be designed to minimize crew errors which could create additional hazards.

Issued in Fort Worth, Texas, on November 13, 2019.

Jorge Castillo,

Manager, Rotorcraft Standards Branch, Policy and Innovation Division, Aircraft Certification Service.

[FR Doc. 2019–25292 Filed 11–20–19; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2019-0894; Product Identifier 2019-NE-32-AD; Amendment 39-19798; AD 2019-21-51]

RIN 2120-AA64

Airworthiness Directives; General Electric Company Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain General Electric Company (GE) GE90-115B model turbofan engines. Emergency AD 2019–21–51 was sent previously to all known operators of the GE GE90–115B model turbofan engines with certain engine serial numbers. This AD requires the removal from service of the Interstage Seal, part number 2505M72P01, from the affected engines. This AD was prompted by a recent event involving an uncontained highpressure turbine (HPT) failure, resulting in an aborted takeoff and debris penetrating the airplane's fuselage and the other engine. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective December 6, 2019 to all persons except those persons to whom it was made immediately effective by Emergency AD 2019–21–51, issued on October 23, 2019, which contained the requirements of this amendment.

The FAA must receive comments on this AD by January 6, 2020.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

• Federal eRulemaking Portal: Go to https://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: 202–493–2251.

• *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

• *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this final rule, contact General Electric Company, GE Aviation, 1 Neumann Way, Cincinnati, OH 45125; phone: 877–432–3272; fax: 877–432–3329; email: aviation.fleetsupport@ge.com. You may view this service information at the FAA, Engine and Propeller Standards Branch, 1200 District Avenue, Burlington, MA, 01803. For information on the availability of this material at the FAA, call 781–238–7759. It is also available on the internet at *https://www.regulations.gov* by searching for and locating Docket No. FAA–2019–0894.

Examining the AD Docket

You may examine the AD docket on the internet at *https:// www.regulations.gov* by searching for and locating Docket No. FAA–2019– 0894; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the regulatory evaluation, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Herman Mak, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781– 238–7147; fax: 781–238–7199; Email: herman.mak@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

On October 23, 2019, the FAA issued Emergency AD 2019–21–51, which requires the removal from service of the GE GE90–115B model turbofan engine Interstage Seal, part number 2505M72P01, from the affected engines. That emergency AD was sent previously to all known operators of GE GE90– 115B model turbofan engines with certain engine serial numbers. This action was prompted by a recent event involving an uncontained HPT failure, that resulted in an aborted takeoff, debris penetrating the airplane's fuselage and the other engine. This condition, if not addressed, could result in uncontained HPT failure, release of high-energy debris, damage to the engine, damage to the airplane, and loss of the airplane.

Related Service Information

The FAA reviewed GE Alert Service Bulletin GE90–100 S/B 72–A0826, dated October 23, 2019. The service information describes procedures for the removal of the Interstage Seal from affected GE GE90–115B model turbofan engines.

FAA's Determination

The FAA is issuing this AD because it evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

AD Requirements

This AD requires the removal from service of the Interstage Seal, part number 2505M72P01, from the affected engines.

FAA's Justification and Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of Emergency AD 2019-21-51, issued on October 23, 2019, to all known U.S. owners and operators of certain GE GE90–115B model turbofan engines. The FAA found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because the Interstage Seal must be removed within 25 flight cycles from the effective date of this AD. Additionally, no domestic operators use the affected engines. The AD is hereby published in the Federal Register as an amendment to section 39.13 of the Federal Aviation Regulations (14 CFR 39.13) to make it effective to all persons. Therefore, the FAA finds good cause that notice and opportunity for prior

public comment are impracticable and unnecessary. In addition, for the reason stated above, the FAA finds that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

This AD is a final rule that involves requirements affecting flight safety and was not preceded by notice and an opportunity for public comment. However, the FAA invites you to send any written data, views, or arguments about this final rule. Send your comments to an address listed under the **ADDRESSES** section. Include the docket number FAA-2019-0894 and Product Identifier 2019-NE-32-AD at the beginning of your comments. The FAA specifically invites comments on the overall regulatory, economic, environmental, and energy aspects of this final rule. The FAA will consider all comments received by the closing date and may amend this final rule because of those comments.

The FAA will post all comments received, without change, to *https:// www.regulations.gov,* including any personal information you provide. The FAA will also post a report summarizing each substantive verbal contact received about this final rule.

Regulatory Flexibility Act

The requirements of the Regulatory Flexibility Act (RFA) do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because FAA has determined that it has good cause to adopt this rule without notice and comment, RFA analysis is not required.

Costs of Compliance

The FAA estimates that this AD affects zero engines installed on airplanes of U.S. registry.

The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Remove the Interstage Seal	100 work-hours × \$85 per hour = \$8,500	\$509,600	\$518,100	\$0

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs describes in more detail the scope of the Agency's authority. The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to engines, propellers, and associated appliances to the Manager, Engine and Propeller Standards Branch, Policy and Innovation Division.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, the FAA certifies that this AD:

(1) Is not a "significant regulatory action" under Executive Order 12866, and

(2) Will not affect intrastate aviation in Alaska.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2019–21–51 General Electric Company: Amendment 39–19798; Docket No. FAA–2019–0894; Product Identifier 2019–NE–32–AD.

(a) Effective Date

This AD is effective December 6, 2019 to all persons except those persons to whom it was made immediately effective by Emergency AD 2019–21–51, issued on October 23, 2019, which contained the requirements of this amendment.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all General Electric Company (GE) GE90–115B model turbofan engines with engine serial numbers 907451, 907464, 907504, 907564, 907574, 907599, 907601, and 907618.

(d) Subject

Joint Aircraft System Component (JASC) Code 7250, Turbine Section.

(e) Unsafe Condition

This AD was prompted by a recent event involving an uncontained high-pressure turbine (HPT) failure, resulting in an aborted takeoff, debris penetrating the aircraft's fuselage and the other engine. The FAA is issuing this AD to prevent failure of the HPT. The unsafe condition, if not addressed, could result in uncontained HPT failure, release of high-energy debris, damage to the engine, damage to the airplane, and loss of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Action

Within 25 flight cycles after the effective date of this AD, remove from service the Interstage Seal, part number 2505M72P01 with serial numbers GWN0TCL3, NCE062LD, NCE254LC, NCE314KU, NCE374LB, NCE527KT, NCE777LD, or NCE994KW.

Note 1 to paragraph (g): GE Alert Service Bulletin GE90–100 S/B 72–A0826, dated October 23, 2019, contains guidance for replacing the Interstage Seal.

(h) Alternative Methods of Compliance (AMOCs)

(1) The Manager, ECO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (i) of this AD. You may email your request to *ANE-AD-AMOC@ faa.gov.*

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/ certificate holding district office.

(i) Related Information

For more information about this AD, contact Herman Mak, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781–238– 7147; fax: 781–238–7199; Email: herman.mak@faa.gov.

(j) Material Incorporated by Reference None.

none

Issued in Burlington, Massachusetts, on November 14, 2019.

Robert J. Ganley,

Manager, Engine and Propeller Standards Branch, Aircraft Certification Service. [FR Doc. 2019–25129 Filed 11–20–19; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF LABOR

Office of Workers' Compensation Programs

20 CFR Part 725

RIN 1240-AA11

Black Lung Benefits Act: Medical Benefit Payments

AGENCY: Office of Workers' Compensation Programs, Labor. **ACTION:** Final rule; delay of applicability date.

SUMMARY: The Office of Workers' Compensation Programs (OWCP) has encountered unforeseen delays in implementing a new computer system to process payments for professional medical and hospital outpatient services made by the Black Lung Disability Trust Fund (Trust Fund) under the Black Lung Benefits Act (BLBA). This action delays the applicability date of two rules relevant to these payments, which were published in the **Federal Register** on June 14, 2018.

DATES:

Effective date: This rule is effective November 21, 2019.

Applicability date: The applicability date for 20 CFR 725.708(a) and (b) and 725.710 is delayed from November 30, 2019 to April 26, 2020.

FOR FURTHER INFORMATION CONTACT:

Michael A. Chance, Director, Division of Coal Mine Workers' Compensation, Office of Workers' Compensation Programs, U.S. Department of Labor, 200 Constitution Avenue NW, Suite N– 3520, Washington, DC 20210. Telephone: 1–800–347–2502. This is a toll-free number. TTY/TDD callers may dial toll-free 1–800–877–8339 for further information.

SUPPLEMENTARY INFORMATION: The Trust Fund pays for covered medical services and treatments provided to certain miners who are entitled to BLBA disability benefits. *See generally* 33 U.S.C. 907, as incorporated by 30 U.S.C. 932(a); 83 FR 27690 (June 14, 2018). On June 14, 2018, OWCP published a final rule revising its regulations governing the payment of medical benefits by the Trust Fund. 83 FR 27690–27699. The revised rules adopt payment formulas that accurately reflect prevailing community rates for authorized treatments and services.

While the revised regulations became effective on August 31, 2018, 83 FR 27690, the Department set a separate applicability date for the rules governing payments for professional medical and outpatient services. *Id.; see*