

item between the LD-MCR compartment and the airplane structure or systems must meet the applicable requirements of § 25.855 at amendment 25-60.

b. Means must be provided to ensure that the fire protection level of the cargo compartment meets the applicable requirements of §§ 25.855 at amendment 25-60; 25.857 at amendment 25-60; and 25.858 at amendment 25-54 when the LD-MCR compartment is not installed.

c. Use of each emergency evacuation route must not require occupants of the LD-MCR compartment to enter the cargo compartment in order to return to the passenger compartment.

d. The aural emergency alarm specified in Special Condition No. 7 must sound in the LD-MCR compartment in the event of a fire in the cargo compartment.

19. Means must be provided to prevent access into the Class C cargo compartment, whether or not the LD-MCR is installed, during all airplane flight operations and to ensure that the maintenance door is closed and secured during all airplane flight operations.

20. All enclosed stowage compartments within the LD-MCR compartment that are not limited to stowage of emergency equipment or airplane supplied equipment (*i.e.*,

bedding), must meet the design criteria given in the table below. As indicated in the table, enclosed stowage compartments larger than 200 ft<sup>3</sup> in interior volume are not addressed by this Special Condition. The in-flight accessibility of very large enclosed stowage compartments and the subsequent impact on the crewmembers' ability to effectively reach any part of the compartment with the contents of a hand fire extinguisher will require additional fire protection considerations similar to those required for inaccessible compartments such as Class C cargo compartments.

| Fire protection features                     | Stowage compartment interior volumes |  |   |
|--|--------------------------------------|--|---|
|  | Less than 25 ft <sup>3</sup>         | 25 ft <sup>3</sup> to 57 ft <sup>3</sup> | 57 ft <sup>3</sup> to 200 ft <sup>3</sup> |
| Materials of Construction <sup>1</sup> ..... | Yes .....                            | Yes .....                                | Yes.                                      |
| Detectors <sup>2</sup> .....                 | No .....                             | Yes .....                                | Yes.                                      |
| Liner <sup>3</sup> .....                     | No .....                             | No .....                                 | Yes.                                      |
| Location Detector <sup>4</sup> .....         | No .....                             | Yes .....                                | Yes.                                      |

<sup>1</sup> *Material*: The material used to construct each enclosed stowage compartment must at least be fire resistant and must meet the flammability standards established for interior components per the requirements of § 25.853. For compartments less than 25 ft<sup>3</sup> in interior volume, the design must ensure the ability to contain a fire likely to occur within the compartment under normal use.

<sup>2</sup> *Detectors*: Enclosed stowage compartments equal to or exceeding 25 ft<sup>3</sup> in interior volume must be provided with a smoke or fire detection system to ensure that a fire can be detected within a one-minute detection time. Flight tests must be conducted to show compliance with this requirement. Each system (or systems) must provide:

(a) A visual indication in the flight-deck within one minute after the start of a fire;  
 (b) An aural warning in the crew rest compartment; and  
 (c) A warning in the main passenger cabin. This warning must be readily detectable by a flight attendant, taking into consideration the positioning of flight attendants throughout the main passenger compartment during various phases of flight.

<sup>3</sup> *Liner*: If it can be shown that the material used to construct the stowage compartment meets the flammability requirements of a liner for a Class B cargo compartment, no liner would be required for enclosed stowage compartments equal to or greater than 25 ft<sup>3</sup> but less than 57 ft<sup>3</sup> in interior volume. For all enclosed stowage compartments equal to or greater than 57 ft<sup>3</sup> but less than or equal to 200 ft<sup>3</sup> in interior volume, a liner must be provided that meets the requirements of § 25.855 at amendment 25-60 for a Class B cargo compartment.

<sup>4</sup> *Location Detector*: LD-MCR compartments that contain enclosed stowage compartments with an interior volume that exceeds 25 ft<sup>3</sup> and are located away from one central location, such as the entry to the LD-MCR compartment or a common area within the LD-MCR compartment, would require additional fire protection features or devices to assist the firefighter in determining the location of a fire.

Issued in Des Moines, Washington, on May 31, 2019.

**Paul Siegmund,**

*Acting Manager, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service.*

[FR Doc. 2019-11957 Filed 6-6-19; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2019-0400; Product Identifier 2019-NM-022-AD]

RIN 2120-AA64

#### Airworthiness Directives; Airbus SAS Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for all Airbus SAS Model A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 airplanes. This proposed AD was prompted by a quality control review, which determined that the wrong aluminum alloy was used to manufacture several structural parts. This proposed AD would require a one-time eddy current conductivity measurement of certain structural parts of the outer flaps to determine if the incorrect alloy was used, and replacement if necessary, as specified in an European Aviation Safety Agency (EASA) AD, which will be incorporated by reference. The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by July 22, 2019.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For the material identified in this proposed AD that will be incorporated by reference (IBR), contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 89990 1000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet [www.easa.europa.eu](http://www.easa.europa.eu). You may find this IBR material on the EASA website at <https://ad.easa.europa.eu>. You may view this IBR material at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available in the AD docket on

the internet at <http://www.regulations.gov>.

### Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0400; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the regulatory evaluation, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

### FOR FURTHER INFORMATION CONTACT:

Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3223.

### SUPPLEMENTARY INFORMATION:

#### Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA–2019–0400; Product Identifier 2019–NM–022–AD” at the beginning of your comments. The FAA specifically invites comments on the overall regulatory, economic, environmental, and energy aspects of this NPRM. The agency will consider all comments received by the closing date and may amend this NPRM based on those comments.

The FAA will post all comments, without change, to <http://www.regulations.gov>, including any personal information you provide. The FAA will also post a report summarizing each substantive verbal contact the agency receives about this NPRM.

#### Discussion

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2019–0012, dated January 24, 2019 (“EASA AD 2019–0012”) (also referred

to as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus SAS Model A321–111, A321–112, A321–131, A321–211, A321–212, A321–213, A321–231 and A321–232 airplanes. The MCAI states:

Following a quality control review on the Airbus final assembly line, it was discovered that wrong aluminum alloy was delivered by a supplier for several structural parts. The results of the investigations highlighted that part of the stock could be impacted by this wrong material. Prompted by this finding, EASA published AD 2015–0218 [which corresponds to FAA AD 2017–05–02 (82 FR 12407, March 3, 2017)], requiring an [special detailed inspection] SDI of certain cabin, cargo compartment and airframe parts. Subsequent investigation results established that outer flaps structure are also affected. Structural investigations demonstrated the capability to sustain the static limits loads, and sufficient fatigue life up to a certain inspection threshold.

This condition, if not detected and corrected, could affect the structural integrity of the outer flap, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, Airbus issued the inspection [service bulletin] SB to provide inspection instructions.

For the reasons described above, this [EASA] AD requires a one-time SDI of suspected parts for material identification and, depending on findings, replacement with serviceable parts.

#### Related IBR Material Under 1 CFR Part 51

EASA AD 2019–0012 describes procedures for a one-time eddy current conductivity measurement of certain structural parts on the outer flaps to determine if an incorrect aluminum alloy was used, and replacement of any affected part with a serviceable part. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the

FAA’s bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is proposing this AD because the agency evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### Proposed Requirements of This NPRM

This proposed AD would require accomplishing the actions specified in EASA AD 2019–0012, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this AD. This proposed AD also would require sending the inspection results to Airbus SAS.

#### Explanation of Required Compliance Information

In the FAA’s ongoing efforts to improve the efficiency of the AD process, the FAA worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. As a result, the FAA expects that EASA AD 2019–0012 will be incorporated by reference in the FAA’s final rule. This proposed AD would, therefore, require compliance with the provisions specified in EASA AD 2019–0012, except for any differences identified as exceptions in the regulatory text of this proposed AD. Service information specified in EASA AD 2019–0012 that is required for compliance with EASA AD 2019–0012 will be available on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0400 after the FAA final rule is published.

#### Costs of Compliance

The FAA estimates that this proposed AD would affect 29 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

#### ESTIMATED COSTS FOR REQUIRED ACTIONS \*

| Labor cost                                 | Parts cost | Cost per product | Cost on U.S. operators |
|--|------------|------------------|------------------------|
| 6 work-hours × \$85 per hour = \$510 ..... | \$0        | \$510            | \$14,790               |

\* Table does not include estimated costs for reporting.

The FAA estimates that it would take about one work-hour per product to

comply with the proposed reporting requirement in this proposed AD. The

average labor rate is \$85 per hour. Based on these figures, the FAA estimates the

cost of reporting the inspection results on U.S. operators to be \$2,465, or \$85 per product.

The FAA has received no definitive data that would enable the agency to provide cost estimates for the on-condition actions specified in this proposed AD.

According to the manufacturer, some or all of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. The FAA does not control warranty coverage for affected individuals. As a result, the agency has included all known costs in the cost estimate.

### Paperwork Reduction Act

A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB control number. The control number for the collection of information proposed by this NPRM is 2120-0056. The paperwork cost associated with this NPRM has been detailed in the Costs of Compliance section of this document and includes time for reviewing instructions, as well as completing and reviewing the collection of information. Therefore, all reporting associated with this NPRM would be mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at 800 Independence Ave. SW, Washington, DC 20591, ATTN: Information Collection Clearance Officer, AES-200.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or

develop on products identified in this rulemaking action.

This proposed AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes and associated appliances to the Director of the System Oversight Division.

### Regulatory Findings

The FAA has determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Will not affect intrastate aviation in Alaska; and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Airbus SAS:** Docket No. FAA-2019-0400; Product Identifier 2019-NM-022-AD.

#### (a) Comments Due Date

The FAA must receive comments by July 22, 2019.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to all Airbus SAS Model A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 airplanes, certificated in any category.

#### (d) Subject

Air Transport Association (ATA) of America Code 57, Wings.

#### (e) Reason

This AD was prompted by a quality control review, which determined that the wrong aluminum alloy was used to manufacture several structural parts. The FAA is issuing this AD to address structural parts made of incorrect aluminum alloy, which could result in reduced structural integrity of the outer flaps and reduced controllability of the airplane.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Aviation Safety Agency (EASA) AD 2019-0012, dated January 24, 2019 ("EASA AD 2019-0012").

#### (h) Exceptions to EASA AD 2019-0012

(1) For purposes of determining compliance with the requirements of this AD: Where EASA AD 2019-0012 refers to its effective date, this AD requires using the effective date of this AD.

(2) The "Remarks" section of EASA AD 2019-0012 does not apply to this AD.

(3) Where paragraph (5) of EASA AD 2019-0012 mandates a parts installation limitation, this AD requires the following parts installation limitation: From the effective date of this AD, only serviceable parts as defined in EASA AD 2019-0012 are allowed to be installed on any airplane.

(4) Where any service information referenced in EASA AD 2019-0012 specifies reporting, this AD requires reporting all inspection results at the applicable time specified in paragraph (h)(4)(i) or (h)(4)(ii) of this AD. If operators have reported findings as part of obtaining any corrective actions approved by Airbus SAS's EASA Design Organization Approval (DOA), operators are not required to report those findings as specified in this paragraph.

(i) If the inspection was done on or after the effective date of this AD: Submit the report within 30 days after the inspection.

(ii) If the inspection was done before the effective date of this AD: Submit the report within 30 days after the effective date of this AD.

#### (i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Section, Transport Standards Branch, FAA,

has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (j)(2) of this AD. Information may be emailed to: [9-ANM-116-AMOC-REQUESTS@faa.gov](mailto:9-ANM-116-AMOC-REQUESTS@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus SAS's EASA DOA. If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* For any service information referenced in EASA AD 2019-0012 that contains RC procedures and tests: Except as required by paragraphs (h)(4) and (i)(2) of this AD, RC procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(4) *Paperwork Reduction Act Burden Statement:* A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately one hour per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW, Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

#### (j) Related Information

(1) For information about EASA AD 2019-0012, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 89990 6017; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); Internet [www.easa.europa.eu](http://www.easa.europa.eu). You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>. You may view this EASA AD at the FAA, Transport Standards

Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. EASA AD 2019-0012 may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2019-0400.

(2) For more information about this AD, contact Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206-231-3223.

Issued in Des Moines, Washington, on May 29, 2019.

**Michael Kaszycki,**

*Acting Director, System Oversight Division, Aircraft Certification Service.*

[FR Doc. 2019-11832 Filed 6-6-19; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2018-0453; Product Identifier 2018-NM-028-AD]

**RIN 2120-AA64**

#### **Airworthiness Directives; Bombardier, Inc., Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Supplemental notice of proposed rulemaking (SNPRM); reopening of comment period.

**SUMMARY:** We are revising an earlier proposal for certain Bombardier, Inc., Model DHC-8-400 series airplanes. This action revises the notice of proposed rulemaking (NPRM) by adding a requirement to replace the lower lock link of the nose landing gear (NLG), which would terminate the repetitive inspections proposed in the NPRM. This action also reduces the applicability in the NPRM. We are proposing this airworthiness directive (AD) to address the unsafe condition on these products. Since these actions would impose an additional burden over those in the NPRM, we are reopening the comment period to allow the public the chance to comment on these changes.

**DATES:** The comment period for the NPRM published in the **Federal Register** on May 30, 2018 (83 FR 24694), is reopened.

We must receive comments on this SNPRM by July 22, 2019.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** 202-493-2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Bombardier, Inc., Q-Series Technical Help Desk, 123 Garratt Boulevard, Toronto, Ontario M3K 1Y5, Canada; telephone: 416-375-4000; fax: 416-375-4539; email: [thd.qseries@aero.bombardier.com](mailto:thd.qseries@aero.bombardier.com); internet: <http://www.bombardier.com>. You may view this referenced service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

#### **Examining the AD Docket**

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2018-0453; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this SNPRM, the regulatory evaluation, any comments received, and other information. The street address for Docket Operations (phone: 800-647-5527) is listed above. Comments will be available in the AD docket shortly after receipt.

#### **FOR FURTHER INFORMATION CONTACT:**

Darren Gassetto, Aerospace Engineer, Mechanical Systems and Administrative Services Section, New York ACO Branch, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7323; fax 516-794-5531.

#### **SUPPLEMENTARY INFORMATION:**

##### **Comments Invited**

We invite you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2018-0453; Product Identifier 2018-NM-028-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy