

workload. Airplane attitudes or crosswind conditions may cause certain symbols (e.g., the zero-pitch line or flight path vector) to reach field-of-view limits such that they cannot be positioned conformally with the image and external scene. In such cases, these symbols may be displayed but with an altered appearance, which makes the pilot aware that they are no longer displayed conformally (for example, "ghosting").

f. A HUD system used to display EFVS images must, if previously certified, continue to meet all of the requirements of the original approval.

3. The safety and performance of the pilot tasks associated with the use of the pilot compartment view must not be degraded by the display of the EFVS image. Pilot tasks that must not be degraded by the EFVS image include:

a. Detection, accurate identification, and maneuvering, as necessary, to avoid traffic, terrain, obstacles, and other hazards of flight.

b. Accurate identification and utilization of visual references required for every task relevant to the phase of flight.

4. Use of EFVS for instrument approach operations must be in accordance with the provisions of § 91.175(l) and (m), and § 121.651, where applicable. Appropriate limitations must be stated in the operating limitations section of the airplane flight manual to prohibit the use of the EFVS for functions that have not been found to be acceptable.

Issued in Renton, Washington, on July 27, 2016.

Victor Wicklund,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016-18445 Filed 8-2-16; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. FAA-2016-7851; Special Conditions No. 25-625-SC]

Special Conditions: Associated Air Center, Boeing Model 747-8 Airplane; Installation of an Airbag System To Limit the Axial Rotation of the Upper Leg on Single-Place Side-Facing Seats

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; request for comments.

SUMMARY: These special conditions are issued for the Boeing Model 747-8 airplane. This airplane, as modified by Associated Air Center, will have a novel or unusual design feature when compared to the state of technology envisioned in the airworthiness standards for transport-category airplanes. This design feature is an airbag system to limit axial rotation of the upper leg, due to leg flail, of occupants in single-place side-facing seats. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for these design features. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: This action is effective on Associated Air Center on August 3, 2016. We must receive your comments by September 19, 2016.

ADDRESSES: Send comments identified by docket number FAA-2016-7851 using any of the following methods:

- *Federal eRegulations Portal:* Go to <http://www.regulations.gov> and follow the online instructions for sending your comments electronically.

- *Mail:* Send comments to Docket Operations, M-30, U.S. Department of Transportation (DOT), 1200 New Jersey Avenue SE., Room W12-140, West Building Ground Floor, Washington, DC 20590-0001.

- *Hand Delivery or Courier:* Take comments to Docket Operations in Room W12-140 of the West Building Ground Floor at 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- *Fax:* Fax comments to Docket Operations at 202-493-2251.

Privacy: The FAA will post all comments it receives, without change, to <http://www.regulations.gov>, including any personal information the commenter provides. Using the search function of the docket Web site, anyone can find and read the electronic form of all comments received into any FAA docket, including the name of the individual sending the comment (or signing the comment for an association, business, labor union, etc.). DOT's complete Privacy Act Statement can be found in the **Federal Register** published on April 11, 2000 (65 FR 19477-19478), as well as at <http://DocketsInfo.dot.gov/>.

Docket: Background documents or comments received may be read at <http://www.regulations.gov> at any time. Follow the online instructions for

accessing the docket or go to Docket Operations in Room W12-140 of the West Building Ground Floor at 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Jayson Claar, FAA, Airframe and Cabin Safety Branch, ANM-115, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue SW., Renton, Washington 98057-3356; telephone 425-227-2194; facsimile 425-227-1320.

SUPPLEMENTARY INFORMATION: The substance of these special conditions has been subject to the public comment process with no comments received. The FAA therefore finds that good cause exists for making these special conditions effective upon publication in the **Federal Register**.

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data.

We will consider all comments we receive by the closing date for comments. We may change these special conditions based on the comments we receive.

Background

On February 15, 2014, Associated Air Center applied for a supplemental type certificate for inflatable airbag systems in the Boeing Model 747-8 airplane. This airplane, currently approved under type certificate no. A20WE, is a private, not-for-hire, not-for-common-carriage business jet with a head-of-state interior. This airplane has a maximum passenger seating capacity of 113. Twelve of the passenger-seating positions include single-place side-facing seats, each of which include an airbag system to protect against leg-flail injuries.

Type Certification Basis

Under the provisions of title 14, Code of Federal Regulations (14 CFR) 21.101, Associated Air Center must show that the Model 747-8 airplane, as changed, continues to meet the applicable provisions of the regulations listed in type certificate no. A20WE, or the applicable regulations in effect on the date of application for the change, except for earlier amendments as agreed upon by the FAA.

If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 25) do not contain

adequate or appropriate safety standards for the Model 747-8 airplane because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions are initially applicable to the model for which they are issued. Should the applicant apply for a supplemental type certificate to modify any other model included on the same type certificate to incorporate the same novel or unusual design feature, these special conditions would also apply to the other model under § 21.101.

In addition to the applicable airworthiness regulations and special conditions, the Boeing Model 747-8 airplane must comply with the fuel-vent and exhaust-emission requirements of 14 CFR part 34 and the noise-certification requirements of 14 CFR part 36.

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type certification basis under § 21.101.

Novel or Unusual Design Features

The Boeing Model 747-8 airplane, as modified by Associated Air Center, will incorporate the following novel or unusual design features: An airbag system to limit axial rotation of the upper leg, due to leg flail, of occupants in single-place side-facing seats.

Discussion

The Boeing Model 747-8 airplane has an interior configuration that includes single-place side-facing seats. These seats include an airbag system in the shoulder belt, per Special Conditions no. 25-606-SC; and an airbag system to limit the axial rotation of the upper leg (femur).

Side-facing seats are considered a novel design for transport-category airplanes that include title 14, Code of Federal Regulations (14 CFR) part 25, Amendment 25-64, in their certification bases and were not anticipated when those airworthiness standards were issued. Therefore, the existing regulations do not provide adequate or appropriate safety standards for occupants of side-facing seats. The FAA issued Special Conditions no. 25-606-SC to address the certification of single- and multiple-place side facing seats for Boeing 747-8 airplanes. Those special conditions include condition 2(e), which requires the axial rotation of the upper-leg (femur) to be limited to 35 degrees in either direction from the nominal seat position. Associated Air Center has developed an airbag system that will be installed close to the floor

and that is designed to limit the axial rotation of the upper-leg.

Serious leg injuries, such as femur fracture, can occur in aviation side-facing seats, injuries that could threaten the occupant's life directly or eliminate the occupant's ability to evacuate the airplane. Limiting upper-leg axial rotation to a conservative limit of 35 degrees (approximately the 50-percentile range of motion) should also limit the risk of serious leg injury. Research suggests that the angle of rotation can be determined by observing lower-leg flailing in typical high-speed video of the dynamic tests. Alternately, the anthropomorphic test dummy could be instrumented to directly measure upper-leg axial rotation. This requirement complies with the intent of the § 25.562(a) injury criteria in preventing serious leg injury.

To comply with special condition 2(e) on some seat positions, Associated Air Center proposes to install leg-flail airbags. This airbag is not addressed in Special Conditions no. 25-606-SC. Therefore, the FAA must issue new special conditions to address this leg-flail airbag installation. These special conditions are similar to other special conditions previously issued for airbags.

Special Conditions no. 25-606-SC for the airbag system in the shoulder belt are based on previous special conditions for airbag systems on forward-facing seat lap belts with some changes to address the specific issues of side-facing seats.

These special conditions for the leg-flail airbag contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

Applicability

As discussed above, these special conditions are applicable to the Boeing Model 747-8 airplane as modified by Associated Air Center. Should Associated Air Center apply at a later date for a supplemental type certificate to modify any other model included on type certificate no. A20WE to incorporate the same novel or unusual design feature, these special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features on one model series of airplanes. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the airplane.

The substance of these special conditions previously has been

subjected to the notice and comment period and has been derived without substantive change from those previously issued. It is unlikely that prior public comment would result in a significant change from the substance contained herein. Therefore, the FAA has determined that prior public notice and comment are unnecessary, and good cause exists for adopting these special conditions upon publication in the **Federal Register**. The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior opportunities for comment described above.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for Boeing Model 747-8 airplanes modified by Associated Air Center.

In addition to the requirements of §§ 25.562 and 25.785, and Special Conditions no. 25-606-SC, the following special conditions are part of the type certification basis for Boeing 747-8 airplanes with leg-flail airbag systems installed on side-facing seats.

1. For seats with leg-flail airbag systems, these systems must deploy and provide protection under crash conditions where it is necessary to prevent serious injury. The means of protection must take into consideration a range of stature from a 2-year-old child to a 95th-percentile male. At some buttock popliteal length and effective seat-bottom depth, the lower legs will not be able to form a 90-degree angle with the upper leg; at this point, the lower-leg flail would not occur. The leg-flail airbag system must provide a consistent approach to prevention of leg flail throughout that range of occupants whose lower legs can form a 90-degree angle relative to the upper legs when seated upright in the seat. Items that need to be considered include, but are not limited to, the range of occupants' popliteal height, the range of occupants' buttock popliteal length, the design of the seat effective height above the floor, and the effective depth of the seat bottom cushion.

2. The leg-flail airbag system must not be susceptible to inadvertent

deployment as a result of wear and tear, or inertial loads resulting from in-flight or ground maneuvers (including gusts and hard landings), and other operating and environmental conditions (vibrations, moisture, etc.) likely to occur in service.

3. Deployment of the leg-flail airbag system must not introduce injury mechanisms to the seated occupant, or result in injuries that could impede rapid egress.

4. Inadvertent deployment of the leg-flail airbag system, during the most critical part of the flight, must either meet the requirement of § 25.1309(b), or not cause a hazard to the airplane or its occupants. This also includes preventing inadvertent airbag deployment from a static discharge.

5. The leg-flail airbag system must not impede rapid egress of occupants from the airplane 10 seconds after airbag deployment.

6. The leg-flail airbag system must be protected from lightning and high-intensity radiated fields (HIRF). The threats to the airplane specified in existing regulations regarding lightning (§ 25.1316) and HIRF (§ 25.1317) are incorporated by reference for the purpose of measuring lightning and HIRF protection.

7. The leg-flail airbag system must function properly after loss of normal airplane electrical power, and after a transverse separation of the fuselage at the most critical location. A separation at the location of the leg-flail airbag system does not have to be considered.

8. The leg-flail airbag system must not release hazardous quantities of gas, sharp injurious metal fragments, or particulate matter into the cabin.

9. The leg-flail airbag system installation must be protected from the effects of fire such that no hazard to occupants will result.

10. A means must be available to verify the integrity of the leg-flail airbag system's activation system prior to each flight, or the leg-flail airbag system's activation system must reliably operate between inspection intervals. The FAA considers that the loss of the leg-flail airbag system's deployment function alone (*i.e.*, independent of the conditional event that requires the leg-flail airbag system's deployment) is a major-failure condition.

11. The airbag inflatable material may not have an average burn rate of greater than 2.5 inches per minute when tested using the horizontal flammability test defined in part 25, appendix F, part I, paragraph (b)(5).

12. The leg-flail airbag system, once deployed, must not adversely affect the emergency-lighting system (*i.e.*, block

floor-proximity lights to the extent that the lights no longer meet their intended function).

Issued in Renton, Washington, on July 26, 2016.

Victor Wicklund,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016-18398 Filed 8-2-16; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-3983; Directorate Identifier 2015-NM-009-AD; Amendment 39-18582; AD 2016-14-01]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; correction.

SUMMARY: The FAA is correcting an airworthiness directive (AD) that published in the **Federal Register**. That AD applies to certain Airbus Model A330-200 Freighter series airplanes; Model A330-200 and A330-300 series airplanes; Model A340-200 and A340-300 series airplanes; Model A340-500 series airplanes; and Model A340-600 series airplanes. Table 1 to paragraph (j) of the regulatory text contains typographical errors regarding certain part numbers (P/Ns). This document corrects those errors. In all other respects, the original document remains the same.

DATES: This final rule is effective August 16, 2016.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of August 16, 2016 (81 FR 44983, July 12, 2016).

ADDRESSES: For Airbus service information identified in this final rule, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 45 80; email airworthiness.A330-A340@airbus.com; Internet <http://www.airbus.com>.

For Hamilton Sundstrand service information identified in this final rule, contact Hamilton Sundstrand, Technical Publications, Mail Stop 302-9, 4747 Harrison Avenue, P.O. Box 7002, Rockford, IL 61125-7002; telephone 860-654-3575; fax 860-998-4564; email

tech.solutions@hs.utc.com; Internet <http://www.hamiltonsundstrand.com>.

You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221. It is also available on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-3983.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1138; fax 425-227-1149.

SUPPLEMENTARY INFORMATION:

Airworthiness Directive 2016-14-01, Amendment 39-18582 (81 FR 44983, July 12, 2016) ("AD 2016-14-01"), currently requires identification of the manufacturer, part number, and serial number of the ram air turbine (RAT), and re-identification and modification of the RAT if necessary, for certain Airbus Model A330-200 Freighter series airplanes; Model A330-200 and A330-300 series airplanes; Model A340-200 and A340-300 series airplanes; Model A340-500 series airplanes; and Model A340-600 series airplanes.

Need for the Correction

As published, table 1 to paragraph (j) of the regulatory text contains typographical errors in two part numbers. Table 1 to paragraph (j) of the AD incorrectly refers to RAT P/Ns 1720934C and 1720934D. Those part numbers should have been 1702934C and 1702934D.

Related Service Information Under 14 CFR Part 51

Airbus has issued the following service information, which describes procedures for identifying the supplier, part number, and serial number of the