

TABLE 1 TO PARAGRAPH (e)—INITIAL INSPECTION THRESHOLDS—Continued

Appendix number of RR ASB Number RB.211–72–AG244, Revision 4, that identifies affected LP compressor blades by S/N	Initial inspection threshold
3G .....	Within 58 months after the effective date of this AD.
3H .....	Within 70 months after the effective date of this AD.
3I .....	Within 82 months after the effective date of this AD.
3J .....	Within 94 months after the effective date of this AD.
3K .....	Within 106 months after the effective date of this AD.
3L .....	Within 118 months after the effective date of this AD.

(2) Thereafter, perform repetitive UIs of the affected LP compressor blades within every 100 flight cycles.

(3) Use paragraph 3.A.(2) of Accomplishment Instructions of RR ASB No. RB.211–72–AG244, Revision 4, dated December 22, 2011, and paragraphs 1. through 3.B. of Appendix 1 of that ASB, or paragraphs 3.B.(3) of Accomplishment Instructions of RR ASB No. RB.211–72–AG244, Revision 4, dated December 22, 2011, and paragraphs 1. through 3.C. of Appendix 2 of that ASB, to perform the UIs. Prior to inspecting the blades per paragraph 3.B.(3) of the Accomplishment Instructions remove the air intake fairing/spinner and spinner extension and annulus fillers.

(4) Do not return to service any engine with blades that failed the inspection required by this AD.

(5) For blades that are removed from the engine and pass inspection, re-apply dry film lubricant before re-installing the blades.

#### (f) Installation Prohibition

After the effective date of this AD, do not install any affected LP compressor blade that has reached the initial inspection threshold in Table 1 to paragraph (e) of this AD, unless it has passed the UI required by this AD.

#### (g) Credit for Previous Actions

You may take credit for the initial inspection that is required by paragraph (e)(1) of this AD if you performed the initial inspection before the effective date of this AD using RR ASB No. RB.211–72–AG244, dated August 7, 2009; RR ASB No. RB.211–72–AG244, Revision 1, dated January 26, 2010; RR ASB No. RB.211–72–AG244, Revision 2, dated August 18, 2011; RR ASB No. RB.211–72–AG244, Revision 3, dated December 13, 2011; or RR RB.211–72–E175, Revision 7, dated April 11, 2011.

#### (h) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request.

#### (i) Related Information

(1) For more information about this AD, contact Alan Strom, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; phone: 781–238–7143; fax: 781–238–7199; email: [alan.strom@faa.gov](mailto:alan.strom@faa.gov).

(2) Refer to European Aviation Safety Agency AD 2012–0025, dated February 8, 2012, for related information.

#### (j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Rolls-Royce plc Alert Service Bulletin No. RB.211–72–AG244, Revision 4, including appendices 1, 2, and 3A through 3L, dated December 22, 2011.

(ii) Reserved.

(3) For service information identified in this AD, contact Rolls-Royce plc, Corporate Communications, P.O. Box 31, Derby, England, DE248BJ, phone: 011–44–1332–242424; fax: 011–44–1332–245418; email: [http://www.rolls-royce.com/contact/civil\\_team.jsp](http://www.rolls-royce.com/contact/civil_team.jsp).

(4) You may view this service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA. For information on the availability of this material at the FAA, call 781–238–7125.

(5) You may view this service information at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: [http://www.archives.gov/federal-register/cfr/ibr\\_locations.html](http://www.archives.gov/federal-register/cfr/ibr_locations.html).

Issued in Burlington, Massachusetts, on August 29, 2012.

**Colleen M. D'Alessandro,**

*Assistant Manager, Engine & Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 2012–23442 Filed 9–21–12; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA–2012–0593; Directorate Identifier 2011–NM–238–AD; Amendment 39–17200; AD 2012–19–05]**

**RIN 2120–AA64**

#### **Airworthiness Directives; Fokker Services B.V. Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain Fokker Services B.V. Model F.28 Mark 0070 and 0100 airplanes. This AD was prompted by reports of burned contacts in a certain production break plug and its corresponding receptacle. This AD requires modifying galley power supply wiring by disconnecting it from the affected plug/receptacle and reconnecting the power supply wiring through splices. We are issuing this AD to prevent a high electrical load, which might lead to overheating of the galley power supply wiring and/or the electrical connector and consequent smoke or fire in the galley area, which could result in damage to the airplane and injury to occupants.

**DATES:** This AD becomes effective October 29, 2012.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 29, 2012.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone (425) 227–1137; fax (425) 425–227–1149.

#### **SUPPLEMENTARY INFORMATION:**

##### **Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on June 12, 2012 (77 FR 34872). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Reports have been received about burned contacts in production break plug P 4259B and corresponding receptacle J 4259A. After investigation, it was concluded that the high

electrical load on the contacts M, L and X, in combination with the electrical loads on the adjacent connections, may have resulted in these occurrences.

This condition, if not detected and corrected, can lead to overheating of the galley power supply wiring and/or the electrical connector and consequent smoke or fire in the galley area, possibly resulting in damage to the aeroplane and injury to occupants.

For the reasons described above, this [European Aviation Safety Agency] AD requires modification of the galley power supply wiring by disconnecting it from the affected plug/receptacle and reconnecting the power supply wiring through splices.

You may obtain further information by examining the MCAI in the AD docket.

### Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM (77 FR 34872, June 12, 2012) or on the determination of the cost to the public.

### Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed—except for minor editorial changes. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM (77 FR 34872, June 12, 2012) for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM (77 FR 34872, June 12, 2012).

### Costs of Compliance

We estimate that this AD will affect 4 products of U.S. registry. We also estimate that it will take about 4 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts will cost about \$210 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$2,200, or \$550 per product.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more

detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM (77 FR 34872, June 12, 2012), the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

#### 2012-19-05 Fokker Services B.V.:

Amendment 39-17200. Docket No. FAA-2012-0593; Directorate Identifier 2011-NM-238-AD.

#### (a) Effective Date

This airworthiness directive (AD) becomes effective October 29, 2012.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to Fokker Services B.V. Model F.28 Mark 0070 and 0100 airplanes, certificated in any category; serial numbers (S/N) 11340 through 11343 inclusive, 11347, 11348, 11350 through 11356 inclusive, 11359, 11360, 11361, 11367 through 11371 inclusive, 11374 through 11378 inclusive, 11382 through 11385 inclusive, 11387 through 11390 inclusive, 11394 through 11397 inclusive, 11400 through 11423 inclusive, 11425 through 11432 inclusive, 11434 through 11439 inclusive, 11441 through 11453 inclusive, and 11456 through 11585 inclusive.

#### (d) Subject

Air Transport Association (ATA) of America Code 24, Electrical Power.

#### (e) Reason

This AD was prompted by reports of burned contacts in a certain production break plug and its corresponding receptacle. We are issuing this AD to prevent a high electrical load, which might lead to overheating of the galley power supply wiring and/or the electrical connector and consequent smoke or fire in the galley area, which could result in damage to the airplane and injury to occupants.

#### (f) Compliance

You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### (g) Modification

Within 24 months after the effective date of this AD: Modify the galley power supply wiring, in accordance with the Accomplishment Instructions of Fokker Service Bulletin SBF100-24-044, dated July 14, 2011, including Fokker Manual Change Notification—Maintenance Documentation MCNM-F100-148, dated July 14, 2011.

#### (h) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone (425) 227-1137; fax (425) 227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### (i) Related Information

Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2011-0183, dated September 23, 2011; and Fokker Service Bulletin SBF100-24-044, dated July 14, 2011, including Fokker Manual Change Notification—Maintenance Documentation MCNM-F100-148, dated July 14, 2011; for related information.

#### (j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Fokker Service Bulletin SBF100-24-044, dated July 14, 2011, including Fokker Manual Change Notification—Maintenance Documentation MCNM-F100-148, dated July 14, 2011.

(ii) Reserved.

(3) For service information identified in this AD, contact Fokker Services B.V., Technical Services Dept., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands; telephone +31 (0)252-627-350; fax +31 (0)252-627-211; email [technicalservices.fokkerservices@stork.com](mailto:technicalservices.fokkerservices@stork.com); Internet <http://www.myfokkerfleet.com>.

(4) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on

the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on September 11, 2012.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

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## DEPARTMENT OF JUSTICE

### Drug Enforcement Administration

#### 21 CFR Part 1300

#### Definitions Relating to Electronic Orders and Prescriptions for Controlled Substances

##### CFR Correction

In Title 21 of the Code of Federal Regulations, Part 1300 to End, revised as of April 1, 2012, on page 14, § 1300.03 is reinstated to read as follows:

#### § 1300.03 Definitions relating to electronic orders for controlled substances and electronic prescriptions for controlled substances.

For the purposes of this chapter, the following terms shall have the meanings specified:

*Application service provider* means an entity that sells electronic prescription or pharmacy applications as a hosted service, where the entity controls access to the application and maintains the software and records on its servers.

*Audit trail* means a record showing who has accessed an information technology application and what operations the user performed during a given period.

*Authentication* means verifying the identity of the user as a prerequisite to allowing access to the information application.

*Authentication protocol* means a well specified message exchange process that verifies possession of a token to remotely authenticate a person to an application.

*Biometric authentication* means authentication based on measurement of the individual's physical features or repeatable actions where those features or actions are both distinctive to the individual and measurable.

*Biometric subsystem* means the hardware and software used to capture, store, and compare biometric data. The biometric subsystem may be part of a larger application. The biometric subsystem is an automated system capable of:

(1) Capturing a biometric sample from an end user.

(2) Extracting and processing the biometric data from that sample.

(3) Storing the extracted information in a database.

(4) Comparing the biometric data with data contained in one or more reference databases.

(5) Determining how well the stored data matches the newly captured data and indicating whether an identification or verification of identity has been achieved.

*Cache* means to download and store information on a local server or hard drive.

*Certificate policy* means a named set of rules that sets forth the applicability of the specific digital certificate to a particular community or class of application with common security requirements.

*Certificate revocation list (CRL)* means a list of revoked, but unexpired certificates issued by a certification authority.

*Certification authority (CA)* means an organization that is responsible for verifying the identity of applicants, authorizing and issuing a digital certificate, maintaining a directory of public keys, and maintaining a Certificate Revocation List.

*Certified information systems auditor (CISA)* means an individual who has been certified by the Information Systems Audit and Control Association as qualified to audit information systems and who performs compliance audits as a regular ongoing business activity.

*Credential* means an object or data structure that authoritatively binds an identity (and optionally, additional attributes) to a token possessed and controlled by a person.

*Credential service provider (CSP)* means a trusted entity that issues or registers tokens and issues electronic credentials to individuals. The CSP may be an independent third party or may issue credentials for its own use.

*CSOS* means controlled substance ordering system.

*Digital certificate* means a data record that, at a minimum—

(1) Identifies the certification authority issuing it;

(2) Names or otherwise identifies the certificate holder;

(3) Contains a public key that corresponds to a private key under the sole control of the certificate holder;

(4) Identifies the operational period; and

(5) Contains a serial number and is digitally signed by the certification authority issuing it.