

(1) Bombardier, Inc. Model CL–600–2C10 (Regional Jet Series 700, 701, & 702) airplanes, serial numbers 10003 through 10331 inclusive.

(2) Bombardier, Inc. Model CL–600–2D15 (Regional Jet Series 705) and CL–600–2D24 (Regional Jet Series 900) airplanes, serial numbers 15001 through 15279 inclusive.

(d) Subject

Air Transport Association (ATA) of America Code 36: Pneumatic.

(e) Reason

This AD was prompted by reports of a bleed air leak from the high pressure ducts which was not immediately detected by the bleed leak detection system. We are issuing this AD to prevent an undetected bleed air leak which can cause loss of rudder control, can lead to degradation of structural integrity, and could be a potential heat source that can lead to fuel being ignited.

(f) Compliance

You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

(g) Install Protective Shields

For Model CL–600–2C10 airplanes having serial numbers 10003 through 10326 inclusive, and Model CL–600–2D15 and CL–600–2D24 airplanes having serial numbers 15001 through 15267 inclusive: Within 6,600 flight hours or 24 months after the effective date of this AD, whichever occurs first, install protective shields on the rudder quadrant support-beam in the aft equipment compartment, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 670BA–36–014, Revision A, dated October 11, 2011.

(h) Install Protective Blankets and Sensing Elements

For Model CL–600–2C10 airplanes having serial numbers 10003 through 10331 inclusive and Models CL–600–2D15 and CL–600–2D24 airplanes having serial numbers 15001 through 15279 inclusive: Within 6,600 flight hours or 24 months after the effective date of this AD, whichever occurs first, install protective blankets on the upper surface of the wing box and fuel components, and install new sensing elements in the wheel well of the main landing gear and the overwing area, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 670BA–36–016, Revision A, dated October 11, 2011.

(i) Credit for Previous Actions

This paragraph provides credit for installations, required by paragraphs (g) and (h) of this AD, if those actions were performed before the effective date of this AD using Bombardier Service Bulletin 670BA–36–014 or 670BA–36–016, both dated April 7, 2011, as applicable.

(j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, New York Aircraft

Certification Office, ANE–170, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the ACO, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone 516–228–7300; fax 516–794–5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product*: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(k) Related Information

Refer to MCAI Canadian Airworthiness Directive CF–2012–06, dated January 26, 2012, and the service bulletins specified in paragraphs (k)(1) and (k)(2) of this AD, for related information.

(1) Bombardier Service Bulletin 670BA–36–014, Revision A, dated October 11, 2011.

(2) Bombardier Service Bulletin 670BA–36–016, Revision A, dated October 11, 2011.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the following service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use the following service information to do the actions required by this AD, unless the AD specifies otherwise.

(i) Bombardier Service Bulletin 670BA–36–014, Revision A, dated October 11, 2011.

(ii) Bombardier Service Bulletin 670BA–36–016, Revision A, dated October 11, 2011.

(3) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514–855–5000; fax 514–855–7401; email thd.crj@aero.bombardier.com; Internet <http://www.bombardier.com>.

(4) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425–227–1221.

(5) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at an NARA facility, call 202–741–6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on May 31, 2012.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2012–14042 Filed 6–15–12; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2012–0562; Directorate Identifier 2012–SW–038–AD; Amendment 39–17068; AD 2012–11–05]

RIN 2120–AA64

Airworthiness Directives; Enstrom Helicopter Corporation Helicopters

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: We are superseding an existing airworthiness directive (AD) for Enstrom Helicopter Corporation (Enstrom) Model F–28C, F–28C–2, F–28F, 280C, 280F, 280FX, TH–28, 480, and 480B helicopters to add another trim relay to the applicability and to revise the modification instructions. This AD is prompted by the discovery that another part-numbered trim relay, inadvertently omitted from the current AD, may contain the same unsafe condition. These actions are intended to prevent failure of the cyclic trim system and subsequent loss of control of the helicopter.

DATES: This AD becomes effective July 3, 2012.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of July 3, 2012.

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of January 23, 2012 (77 FR 729, January 6, 2012).

We must receive comments on this AD by August 17, 2012.

ADDRESSES: You may send comments by any of the following methods:

- *Federal eRulemaking Docket*: Go to <http://www.regulations.gov>. Follow the online instructions for sending your comments electronically.

- *Fax*: 202–493–2251.

- *Mail*: Send comments to the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200

New Jersey Avenue SE., Washington, DC 20590-0001.

• **Hand Delivery:** Deliver to the "Mail" address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the economic evaluation, any comments received, and other information. The street address for the Docket Operations Office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

For service information identified in this AD, contact Enstrom Helicopter Corporation, 2209 22nd St., Menominee, Michigan, 49858-0490; telephone: 906-863-1200; email: customerservice@enstromhelicopter.com; Web site: http://www.enstromhelicopter.com/enstrom_new/enstrom_support_tec.html. You may review a copy of the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

FOR FURTHER INFORMATION CONTACT:

Gregory J. Michalik, Enstrom Program Manager, FAA, Chicago Aircraft Certification Office, 2300 East Devon Avenue, Room 107, Des Plaines, Illinois, 60018; telephone (847) 294-7135; email: gregory.michalik@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not provide you with notice and an opportunity to provide your comments prior to it becoming effective. However, we invite you to participate in this rulemaking by submitting written comments, data, or views. We also invite comments relating to the economic, environmental, energy, or federalism impacts that resulted from adopting this AD. The most helpful comments reference a specific portion of the AD, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit them only one time. We will file in the docket all comments that we receive, as well as a report summarizing each substantive public contact with

FAA personnel concerning this rulemaking during the comment period. We will consider all the comments we receive and may conduct additional rulemaking based on those comments.

Discussion

On December 14, 2011, we issued AD 2011-26-10, amendment 39-16900 (77 FR 729, January 6, 2012) for Enstrom Model F-28C, F-28C-2, F-28F, 280C, 280F, 280FX, TH-28, 480, and 480B helicopters with a trim relay, part-number (P/N) KUP14D55-472, M83536/10-015M, or M83536/10-024M. That AD requires modifying the lateral and longitudinal trim actuator assemblies by replacing the actuator and limit switch bracket to provide a positive stop for the trim actuator. In the event of a trim actuator runaway, the new bracket will stop the actuator, causing the circuit breaker to trip before any significant loss of control occurs. After the trim actuator assemblies are modified, that AD requires performing operational (ground) and flight tests to determine that the trim relay is working correctly. That AD was prompted by reports of 4 failures in the cyclic trim system in the field, 2 that occurred on the Enstrom Model 480B helicopter and 2 that occurred on the Enstrom Model F28 helicopter. These failures resulted in reduced controllability of the helicopter. We issued that AD to prevent failure of the cyclic trim system and subsequent loss of control of the helicopter.

Comments

Since we issued immediately adopted AD 2011-26-10, we received one comment from Enstrom, the manufacturer, and Revision 4 to Enstrom Service Directive Bulletin No. 0110, dated January 23, 2012 (SDB 0110), for Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters. In its comment, the manufacturer stated that the list of affected part numbers for the F-28 series and 280 series helicopter should include P/N KUP14D55-24 and that this P/N was erroneously left off of Enstrom's original Service Directive Bulletin. It stated that the same safety concerns also exist with this relay and that adding this P/N will not affect the cost analysis or fleet size as these aircraft were included in the initial calculations.

We agree with the commenter that the additional part numbered relay contained in Enstrom SDB 0110 should be added to the applicability section of the AD because it may contain the same unsafe condition. Also, we agree that the revised compliance instructions contained in SDB 0110 should be required.

FAA's Determination

We are issuing this AD because we evaluated all the relevant information, including the comment received and the revised service information, and determined the unsafe condition described previously is likely to exist or develop in other helicopters of these same type designs.

Related Service Information

We reviewed Enstrom SDB No. T-039, Revision 3, dated July 6, 2011 (SDB T-039), for Model TH-28, 480, and 480B helicopters and Enstrom SDB 0110, for Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters. SDB T-039 specifies, for helicopters with a trim relay, P/N M83536/10-024M, procedures for modifying the lateral and longitudinal trim actuator assembly using the cyclic trim assembly kit (modification kit), P/N 4230045-1, and specifies performing an operational check and flight test to determine if the trim is operating correctly after the modification. SDB 0110 specifies, for helicopters with a trim relay, P/N KUP14D55-24, KUP14D55-472, M83536/10-015M, or M83536/10-024M, procedures for modifying the lateral and longitudinal trim actuator assembly using the modification kit, P/N 28-01063-1, and specifies performing an operational check and flight test to determine if the trim is operating correctly after the modification. The SDBs state that the modification kits contain the upgraded bracket.

AD Requirements

In addition to those part-numbered trim relays contained in the applicability section of the current AD 2011-26-10, this AD adds trim relay P/N KUP14D55-24. Also, this AD adopts the revised modification instructions for the Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters contained in SDB 0110. Otherwise, this AD retains most of the requirements of AD 2011-26-10, which requires modifying the lateral and longitudinal trim actuator assembly, using modification kit, P/N 28-01063-1, for the Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters and modification kit, P/N 4230045-1, for the Model TH-28, 480, and 480B helicopters, and verifying that the modification has been completed properly by performing an operational and flight test after the modification is complete.

The actions required by this AD are to be accomplished by following specified

portions of the SDBs described previously.

Differences Between This AD and the Service Information

The SDBs specify, before further flight, to insert a special addendum into the Emergency Procedures section of the Flight Manual, and this AD does not require this action.

Costs of Compliance

We estimate that this AD will affect 207 helicopters of U.S. Registry. We estimate that operators may incur the following costs in order to comply with this AD. For Model 480, 480B, and TH-28, modifying the actuator assemblies will require 8 work hours at a cost of \$85 per hour and parts will cost \$327; the cost per helicopter will be \$1,007. For Model 280C, 280F, 280FX, F-28C, F-28C-2, and F-28F, modifying the actuator assemblies will require 8 work hours at a cost of \$85 per hour; parts will cost \$383; and the cost per helicopter will be \$1,063.

FAA's Justification and Determination of the Effective Date

Providing an opportunity for public comments prior to adopting these AD requirements would delay implementing the safety actions needed to correct this known unsafe condition. Therefore, we find that the risk to the flying public justifies waiving notice and comment prior to the adoption of this rule because the required corrective actions must be accomplished within 5 hours time-in-service, a very short time period based on the *average* flight-hour utilization rate of these helicopters.

Since an unsafe condition exists that requires the immediate adoption of this AD, we determined that notice and opportunity for public comment before issuing this AD are impracticable and contrary to the public interest and that good cause exists for making this amendment effective in less than 30 days.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations

for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared an economic evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Amendment 39-16900 (77 FR 729, January 6, 2012), and adding the following new airworthiness directive (AD):

2012-11-05 Enstrom Helicopter

Corporation: Amendment 39-17068; Docket No. FAA-2012-0562; Directorate Identifier 2012-SW-038-AD.

(a) Applicability

This AD applies to the Enstrom Model F-28C, F-28C-2, F-28F, 280C, 280F, 280FX,

TH-28, 480, and 480B helicopters with a trim relay, part-number (P/N) KUP14D55-24, KUP14D55-472, M83536/10-015M, or M83536/10-024M, certificated in any category.

Note to paragraph (a) of this AD: This AD does not apply to the specified helicopters with a reversible trim motor, P/N 28-16621 (Ford Motor Company C1AZ-14553A) or P/N AD1R-10 (Signal Electric).

(b) Unsafe Condition

This AD defines the unsafe condition as a failure of a trim relay in the cyclic trim system. This condition could result in reduced controllability of the helicopter and subsequent loss of control of the helicopter.

(c) Other Affected ADs

This AD supersedes AD 2011-26-10, Amendment 39-16900 (77 FR 729, January 6, 2012).

(d) Effective Date

This AD becomes effective July 3, 2012.

(e) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(f) Required Actions

Within 5 hours time-in-service (TIS) or at the next annual or 100 hour TIS inspection, whichever occurs first:

(1) For the Enstrom Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters, modify the lateral and longitudinal trim actuator assemblies using the cyclic trim assembly kit (modification kit), P/N 28-01063-1, in accordance with the instructions in paragraph 6.1 of the Enstrom Service Directive Bulletin (SDB) No. 0110, Revision 4, dated January 23, 2012 (SDB No. 0110 R4), except when the instructions specify using "Aeroshell 22 grease" or "VC-3 Vibra-tite thread locker," you may use an equivalent product.

(2) For the Enstrom Model TH-28, 480, and 480B helicopters, modify the lateral and longitudinal trim actuator assemblies using the modification kit, P/N 4230045-1, in accordance with the instructions in paragraph 6.1 of the Enstrom SDB No. T-039, Revision 3, dated July 6, 2011 (SDB No. T-039 R3), except when the instructions specify using "Aeroshell 22 grease" or "VC-3 Vibra-tite thread locker," you may use an equivalent product, and you are not required to contact Enstrom Customer Service.

(3) After modifying the lateral and longitudinal trim actuator assemblies in accordance with paragraphs (f)(1) or (f)(2) of this AD, before further flight, operationally test the trim limits in accordance with paragraph 6.2. of the SDB for your model helicopter, and determine during a flight test whether there is appropriate trim authority in accordance with paragraph 6.3. of the SDB for your model helicopter.

(g) Special Flight Permits

A one-time special-flight permit may be issued in accordance with 14 CFR 21.197 and 21.199 provided the helicopter is operated with the trim system circuit breaker pulled.

(h) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Chicago Aircraft Certification Office, FAA, may approve AMOCs for this AD. Send your proposal to: Gregory J. Michalik, Enstrom Program Manager, FAA, Chicago Aircraft Certification Office, 2300 East Devon Avenue, Room 107, Des Plaines, Illinois, 60018; telephone (847) 294-7135; fax (847) 294-7834; email: gregory.michalik@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

(i) Subject

Joint Aircraft Service Component (JASC) Code: 6710: Main Rotor Control.

(j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the following service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use the following service information to do the actions required by this AD, unless the AD specifies otherwise.

(3) The following service information was approved for IBR on July 3, 2012.

(i) Enstrom Service Directive Bulletin No. 0110, Revision 4, dated January 23, 2012, for Model F-28C, F-28C-2, F-28F, 280C, 280F, and 280FX helicopters.

(4) The following service information was approved for IBR on January 23, 2012 (77 FR 729, January 6, 2012).

(i) Enstrom Service Directive Bulletin No. T-039, Revision 3, dated July 6, 2011, for Model TH-28, 480, and 480B helicopters.

(5) For service information identified in this AD, contact Enstrom Helicopter Corporation, 2209 22nd St., Menominee, Michigan, 49858-0490; telephone: 906-863-1200; email: customerservice@enstromhelicopter.com; Web site: http://www.enstromhelicopter.com/enstrom_new/enstrom_support_tec.html.

(6) You may review a copy of the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137 or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal-register/code_of_federal_regulations/ibr_locations.html.

Issued in Fort Worth, Texas, on May 17, 2012.

Kim Smith,

Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 2012-14634 Filed 6-15-12; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2011-1254; Directorate Identifier 2010-NM-178-AD; Amendment 39-17083; AD 2012-12-04]

RIN 2120-AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are superseding an existing airworthiness directive (AD) for certain The Boeing Company Model 737-300, -400, and -500 series airplanes. That AD currently requires repetitive external detailed inspections or non-destructive inspections to detect cracks in the fuselage skin along the chem-mill steps at stringers S-1 and S-2R, between station (STA) 400 and STA 460, and repair if necessary. This new AD adds inspections for cracking in additional fuselage skin locations, and repair if necessary. This new AD also reduces the inspection thresholds and repetitive intervals for certain airplanes. This AD was prompted by reports of additional crack findings of the fuselage skin at the chem-mill steps. We are issuing this AD to detect and correct fatigue cracking of the fuselage skin panels at the chem-mill steps, which could result in sudden fracture and failure of the fuselage skin panels, and consequent rapid decompression of the airplane.

DATES: This AD is effective July 23, 2012.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of July 23, 2012.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the

Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Wayne Lockett, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, Washington 98057-3356; phone: 425-917-6447; fax: 425-917-6590; email: wayne.lockett@faa.gov.

SUPPLEMENTARY INFORMATION:**Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2008-19-03, Amendment 39-15670 (73 FR 56958, October 1, 2008). That AD applies to the specified products. The NPRM published in the **Federal Register** on November 28, 2011 (76 FR 72853). That NPRM proposed to continue to require repetitive external detailed inspections or non-destructive inspections to detect cracks in the fuselage skin along the chem-mill steps at stringers S-1 and S-2R, between station (STA) 400 and STA 460, and repair if necessary. That NPRM also proposed to add inspections for cracking in additional fuselage skin locations, and repair if necessary. That NPRM also proposed to reduce the inspection thresholds and repetitive intervals for certain airplanes.

Comments

We gave the public the opportunity to participate in developing this AD. The following presents the comments received on the proposal (76 FR 72853, November 28, 2011) and the FAA's response to each comment.

Support for the NPRM (76 FR 72853, November 28, 2011)

The National Transportation Board supports the NPRM (76 FR 72853, November 28, 2011).

Request To Revise Paragraph (j) of the NPRM (76 FR 72853, November 28, 2011)

Boeing asked that we include Boeing Alert Service Bulletin 737-53A1293, Revision 1, dated July 7, 2010, in the exception to the service bulletin