

sources and devices that are registered under the § 32.210 process;

- Under § 34.20(b)(3), a licensee is allowed to modify associated equipment unless the design of any replacement component would compromise the design safety features of the system;

- A licensee's modification process must account for prototype testing or engineering analysis of a replacement component against the performance criteria required in § 34.20 for any component that was modified for use in licensed activities;

- To comply with § 34.20, a licensee should demonstrate that modifications to associated equipment: (1) Will not create material incompatibility that may degrade a source or device over their expected useful life times; (2) will not diminish the performance of associated equipment in expected use environments over the expected life time of the associated equipment; (3) will not allow a source to inadvertently exit the system; and (4) will not compromise expected safe use of the system; and

- Enforcement action would be considered for a licensee who completes modification of associated equipment that compromises the design safety features of the system. The NRC Enforcement Policy (NUREG-1600) includes an example involving possession or use of unauthorized equipment which degrades safety in the conduct of licensee activities.

The NRC has determined that alignment of the NRC implementation to the existing NRC requirements maintains the same level of compatibility between the Agreement State regulations and the existing NRC requirements. Also, use of revised NRC guidance rather than rulemaking to achieve the petitioner's intent provides Agreement States the flexibility to revise their policy and guidance to meet unique situations and local conditions.

In conclusion, no new information has been provided by the petitioner that calls into question the requirements. Existing NRC regulations provide the basis for reasonable assurance that the common defense and security and public health and safety are adequately protected; therefore, rulemaking does not appear to be warranted.

For the reasons cited in this document, the NRC denies this petition.

Dated at Rockville, Maryland, this 9th day of July, 2003.

For the Nuclear Regulatory Commission.

Annette Vietti-Cook,
Secretary of the Commission.

[FR Doc. 03-17846 Filed 7-14-03; 8:45 am]

BILLING CODE 7590-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-CE-26-AD]

RIN 2120-AA64

Airworthiness Directives; GROB-WERKE Model G120A Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to all GROB-WERKE (GROB) Model G120A airplanes. This proposed AD would require you to modify the flight control system operating levers. This proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions specified by this proposed AD are intended to prevent failure of a ball bearing in flight control system operating levers. Such failure could lead to reduced control or loss of control of the airplane.

DATES: The Federal Aviation Administration (FAA) must receive any comments on this proposed rule on or before August 18, 2003.

ADDRESSES: Submit comments to FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003-CE-26-AD, 901 Locust, Room 506, Kansas City, Missouri 64106. You may view any comments at this location between 8 a.m. and 4 p.m., Monday through Friday, except Federal holidays. You may also send comments electronically to the following address: 9-CE-7-Docket@faa.gov. Comments sent electronically must contain "Docket No. 2003-CE-26-AD" in the subject line. If you send comments electronically as attached electronic files, the files must be formatted in Microsoft Word 97 for Windows or ASCII text.

You may get service information that applies to this proposed AD from GROB Luft-und Raumfahrt, Lettenbachstrasse 9, D-86874 Tussenhausen-Mattsies, Germany; telephone: 011 49 8268 998139; facsimile: 011 49 8268 998200; email: productsupport@grob-aerospace.de. You may also view this information at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106;

telephone: (816) 329-4146; facsimile: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Comments Invited

How Do I Comment on This Proposed AD?

The FAA invites comments on this proposed rule. You may submit whatever written data, views, or arguments you choose. You need to include the proposed rule's docket number and submit your comments to the address specified under the caption **ADDRESSES**. We will consider all comments received on or before the closing date. We may amend this proposed rule in light of comments received. Factual information that supports your ideas and suggestions is extremely helpful in evaluating the effectiveness of this proposed AD action and determining whether we need to take additional rulemaking action.

Are There Any Specific Portions of This Proposed AD I Should Pay Attention To?

The FAA specifically invites comments on the overall regulatory, economic, environmental, and energy aspects of this proposed rule that might suggest a need to modify the rule. You may view all comments we receive before and after the closing date of the rule in the Rules Docket. We will file a report in the Rules Docket that summarizes each contact we have with the public that concerns the substantive parts of this proposed AD.

How Can I Be Sure FAA Receives My Comment?

If you want FAA to acknowledge the receipt of your mailed comments, you must include a self-addressed, stamped postcard. On the postcard, write "Comments to Docket No. 2003-CE-26-AD." We will date stamp and mail the postcard back to you.

Discussion

What Events Have Caused This Proposed AD?

The Luftfahrt-Bundesamt (LBA), which is the airworthiness authority for Germany, recently notified FAA that an unsafe condition may exist on all GROB Model G120A airplanes. The LBA reports that a damaged ball bearing in a flight control system operating lever was found. The damage was found during regular maintenance. The damage is believed to be caused by incorrect installation.

What Are the Consequences if the Condition Is Not Corrected?

If not corrected, this condition could cause failure of a ball bearing in affected flight control system operating levers. Such failure could result in reduced control or loss of control of the airplane.

Is There Service Information That Applies to This Subject?

GROB has issued Service Letter No. SL1121-009, dated May 23, 2003; Service Bulletin No. MSB1121-033, dated May 8, 2003; and Service Bulletin No. MSB1121-034, dated May 19, 2003.

What Are the Provisions of This Service Information?

GROB Service Bulletin No. MSB1121-033, dated May 8, 2003, includes procedures for inspecting all flight control system operating levers for damaged ball bearings and replacing any lever that has a damaged ball bearing.

GROB Service Bulletin No. MSB-1121-034, dated May 19, 2003, includes procedures for modifying the flight control system operating levers.

GROB Service Letter No. SL1121-009, dated May 23, 2003, includes procedures for modifying elevator rod 1.

What Action Did the LBA Take?

The LBA classified these service bulletins as mandatory and issued German AD Number 2003-164/2, dated

May 22, 2003, in order to ensure the continued airworthiness of these airplanes in Germany.

Was This in Accordance With the Bilateral Airworthiness Agreement?

This airplane model is manufactured in Germany and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement.

Pursuant to this bilateral airworthiness agreement, the LBA has kept FAA informed of the situation described above.

The FAA's Determination and an Explanation of the Provisions of This Proposed AD

What Has FAA Decided?

The FAA has examined the findings of the LBA; reviewed all available information, including the service information referenced above; and determined that:

- The unsafe condition referenced in this document exists or could develop on other GROB Model G120A airplanes of the same type design that are on the U.S. registry;
- The actions specified in the previously-referenced service information should be accomplished on the affected airplanes; and

—AD action should be taken in order to correct this unsafe condition.

What Would This Proposed AD Require?

This proposed AD would require you to incorporate the actions in the previously-referenced service information.

How Does the Revision to 14 CFR Part 39 Affect This Proposed AD?

On July 10, 2002, FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs FAA's AD system. This regulation now includes material that relates to special flight permits, alternative methods of compliance, and altered products. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Cost Impact

How Many Airplanes Would This Proposed AD Impact?

We estimate that this proposed AD affects 6 airplanes in the U.S. registry.

What Would be the Cost Impact of This Proposed AD on Owners/Operators of the Affected Airplanes?

We estimate the following costs to accomplish these proposed modifications:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. operators
10 workhours × \$60 per hour = \$600	No cost for parts	\$600	6 × \$600 = \$3,600

Regulatory Impact

Would This Proposed AD Impact Various Entities?

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposed rule would not have federalism implications under Executive Order 13132.

Would This Proposed AD Involve a Significant Rule or Regulatory Action?

For the reasons discussed above, I certify that this proposed action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant

economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends § 39.13 by adding a new airworthiness directive (AD) to read as follows:

Grob-Werke: Docket No. 2003-CE-26-AD.

(a) *What airplanes are affected by this AD?* This AD affects Model G120A airplanes, all serial numbers, that are certificated in any category.

(b) *Who must comply with this AD?* Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?* The actions specified by this AD are intended to prevent failure of a ball bearing in flight control system operating levers. Such failure could lead to reduced control or loss of control of the airplane.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Inspect the flight control system operating levers for damaged ball bearings and replace any lever with a damaged ball bearing.	Inspect within the next 50 hours time-in-service (TIS) after the effective date of this AD. Replace prior to further flight after the inspection.	In accordance with GROB Service Bulletin No. MSB1121-033, dated May 8, 2003.
(2) Accomplish the modifications to: (a) elevator rod 1, part number (P/N) 120A-4400.08 or part number 120A-4217 (which supersedes P/N 120A-4400.08); and (b) the flight control system operating levers	Within the next 50 hours TIS after the effective date of this AD.	In accordance with GROB Service Letter No. SL1121-009, dated May 23, 2003, and GROB Service Bulletin No. MSB1121-034, dated May 19, 2003.
(3) Only install flight control system operating levers that have been modified in accordance with paragraph (d)(2)(a) and (d)(2)(b) of this AD.	As of the effective date of this AD	

(e) *Can I comply with this AD in any other way?* To use an alternative method of compliance or adjust the compliance time, follow the procedures in 14 CFR 39.19. Send these requests to the Manager, Standards Office, Small Airplane Directorate. For information on any already approved alternative methods of compliance, contact Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4146; facsimile: (816) 329-4090.

(f) *How do I get copies of the documents referenced in this AD?* You may get copies of the documents referenced in this AD from GROB Luft-und Raumfahrt, Lettenbachstrasse 9, D-86874 Tussenhausen-Mattsies, Germany; telephone: 011 49 8268 998139; facsimile: 011 49 8268 998200; email: productssupport@grob-aerospace.de. You may view these documents at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106.

Note: The subject of this AD is addressed in German AD 2003-164/2, dated May 22, 2003.

Issued in Kansas City, Missouri, on July 9, 2003.

Dorenda D. Baker,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-17818 Filed 7-14-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NM-319-AD]

RIN 2120-AA64

Airworthiness Directives; Dornier Model 328-300 Series Airplanes Equipped With Certain Pratt & Whitney PW306B Engine Nacelles

AGENCY: Federal Aviation Administration, DOT.

ACTION: Supplemental notice of proposed rulemaking; reopening of comment period.

SUMMARY: This document revises an earlier proposed airworthiness directive (AD), applicable to certain Dornier Model 328-300 series airplanes, that would have required performing a check of the airplane maintenance records; inspecting the engine nacelle anti-ice tube for leaks, if necessary; and modifying the joint, if necessary. This new action revises the proposed rule by removing the requirement to perform a records check, which was intended to allow operators to determine whether the inspection would be required. The actions specified by this new proposed AD are intended to prevent an uncommanded engine shutdown in a critical phase of flight due to leakage of air from a loose clamp on the anti-ice tubing joint. This action is intended to address the identified unsafe condition. **DATES:** Comments must be received by August 11, 2003.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-

319-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2001-NM-319-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 or 2000 or ASCII text.

The service information referenced in the proposed rule may be obtained from FAIRCHILD Dornier GmbH, P.O. Box 1103, D-82230 Wessling, Germany. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1175; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained