Action	Compliance time	Procedures
(2) You may not install,on any affected airplane, a nose landing gear unit that does not incorporate a modified steering actuator piston rod, as required by paragraph (d)(1) of this AD.		Not Applicable.

- (e) Can I comply with this AD in any other way? You may use an alternative method of compliance or adjust the compliance time if:
- (1) Your alternative method of compliance provides an equivalent level of safety; and
- (2) The Manager, Small Airplane
 Directorate, approves your alternative.
 Submit your request through an FAA
 Principal Maintenance Inspector, who may
 add comments and then send it to the
 Manager, Small Airplane Directorate.

Note 1: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if you have not eliminated the unsafe condition, specific actions you propose to address it.

- (f) Where can I get information about any already-approved alternative methods of compliance? Contact Mr. S.M. Nagarajan, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4145; facsimile: (816) 329–4090.
- (g) What if I need to fly the airplane to another location to comply with this AD? The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.
- (h) How do I get copies of the documents referenced in this AD? You may obtain copies of the documents referenced in this AD from British Aerospace Regional Aircraft, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland. You may examine these documents at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106.

Note 2: The subject of this AD is addressed in British AD 004–05–2000.

Issued in Kansas City, Missouri, on October 17, 2000.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00–27223 Filed 10–23–00; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-CE-83-AD]

RIN 2120-AA64

Airworthiness Directives; British Aerospace HP137 Mk1, Jetstream Series 200, and Jetstream Models 3101 and 3201 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to certain British Aerospace HP137 Mk1, Jetstream series 200, and Jetstream Models 3101 and 3201 airplanes that are equipped with certain nose landing gear units. The proposed AD would require you to inspect the steering jack assembly to assure proper clearance between the bush heads on the steering plates and the shim on the steering jack trunnions and to assure that there is adequate lubrication at both trunnions and the eve end fitting. The proposed AD would also require you to adjust the clearance and provide adequate lubrication, as necessary. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for the United Kingdom. The actions specified by the proposed AD are intended to prevent cracked steering jack piston rods caused by inadequate clearance or inadequate lubrication of the steering jack pivot points. The condition could result in failure of the nose wheel steering system with consequent loss of airplane control.

DATES: The Federal Aviation Administration (FAA) must receive any comments on this proposed rule on or before November 22, 2000.

ADDRESSES: Submit comments in triplicate to FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 99–CE–83–AD, 901

Locust, Room 506, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from British Aerospace Regional Aircraft, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland; telephone: (01292) 479888; facsimile: (01292) 479703. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. S.M. Nagarajan, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–

4145; facsimile: (816) 329–4090.

SUPPLEMENTARY INFORMATION:

Comments Invited

How do I comment on the proposed *AD?* The FAA invites comments on this proposed rule. You may submit whatever written data, views, or arguments you choose. You need to include the rule's docket number and submit your comments in triplicate to the address specified under the caption ADDRESSES. The FAA will consider all comments received on or before the closing date. We may amend the proposed rule in light of comments received. Factual information that supports your ideas and suggestions is extremely helpful in evaluating the effectiveness of the proposed AD action and determining whether we need to take additional rulemaking action.

Are there any specific portions of the proposed AD I should pay attention to? The FAA specifically invites comments on the overall regulatory, economic, environmental, and energy aspects of the proposed rule that might suggest a need to modify the rule. You may examine all comments we receive before and after the closing date of the rule in the Rules Docket. We will file a report in the Rules Docket that summarizes each FAA contact with the public that

concerns the substantive parts of the proposed AD.

We are re-examining the writing style we currently use in regulatory documents, in response to the Presidential memorandum of June 1, 1998. That memorandum requires Federal agencies to communicate more clearly with the public. We are interested in your comments on whether the style of this document is clearer, and any other suggestions you might have to improve the clarity of FAA communications that affect you. You can get more information about the Presidential memorandum and the plain language initiative at http:// www.plainlanguage.gov.

How can I be sure FAA receives my comment? If you want us to acknowledge the receipt of your comments, you must include a selfaddressed, stamped postcard. On the postcard, write "Comments to Docket No. 99–CE–83–AD.'' We will date stamp and mail the postcard back to you.

Discussion

What events have caused this proposed AD? The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom, recently notified the FAA that an unsafe condition may exist on certain British Aerospace HP137 Mk1, Jetstream series 200, and Jetstream Models 3101 and 3201 airplanes. The CAA reports incidents where cracking has occurred at the base of the thread in the steering jack piston rod, part number 618212, in the nose landing gear unit. The condition could occur on the referenced airplanes that are equipped with nose landing gear unit 1873, B00A702852A, B00A703064A, or B00A703056A.

Inadequate clearance or inadequate lubrication of the steering jack pivot points can result in unusually high operational loads. These loads could result in such cracks in the steering jack

piston rod.

What are the consequences if the condition is not corrected? A cracked steering jack piston rod could result in failure of the nose wheel steering system with consequent loss of airplane

Is there service information that applies to this subject? British Aerospace has issued Mandatory Service Bulletin 32-JA 981043, dated March 5, 1999. This service bulletin:

- Specifies inspecting the steering jack assembly to assure proper clearance between the bush heads on the steering plates and the shim on the steering jack trunnions and to assure that there is adequate lubrication at both trunnions and the eye end fitting;
- · Specifies adjusting the clearance and providing adequate lubrication, as necessary; and
- References APPH Ltd. Service Newsletter, Issue 2, Jetstream 31 Steering Jack Part Number 618200. The procedures to accomplish the abovereferenced actions are included in this service newsletter.

What action did the CAA take? The CAA classified this service bulletin as mandatory and issued British AD Number 012–03–99, in order to assure the continued airworthiness of these airplanes in the United Kingdom.

Was this in accordance with the bilateral airworthiness agreement? These airplane models are manufactured in the United Kingdom and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement.

Pursuant to this bilateral airworthiness agreement, the CAA has kept FAA informed of the situation described above.

The FAA's Determination and an **Explanation of the Provisions of the** Proposed AD

What has FAA decided? The FAA has examined the findings of the CAA; reviewed all available information, including the service information referenced above; and determined that:

- The unsafe condition referenced in this document exists or could develop on other British Aerospace HP137 Mk1, Jetstream series 200, and Jetstream Models 3101 and 3201 airplanes of the same type design;
- The actions specified in the referenced service information should be accomplished on the affected airplanes; and
- · AD action should be taken in order to correct this unsafe condition.

What would the proposed AD require? This proposed AD would require you to inspect the steering jack assembly to assure proper clearance between the bush heads on the steering plates and the shim on the steering jack trunnions and to assure that there is adequate lubrication at both trunnions and the eye end fitting. The proposed AD would also require you to adjust the clearance and provide adequate lubrication, as necessary.

Accomplishment of the proposed action would be in accordance with APPH Ltd. Service Newsletter, Issue 2, Jetstream 31 Steering Jack Part Number 618200, as referenced in British Aerospace Mandatory Service Bulletin 32-JA 981043, dated March 5, 1999.

Cost Impact

How many airplanes would the proposed AD impact? We estimate that the proposed AD would affect 300 airplanes in the U.S. registry.

What would be the cost impact of the proposed AD on owners/operators of the affected airplanes? We estimate the following costs to accomplish the proposed inspection:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. airplane operators
1 workhour × \$60 per hour	No parts required to accomplish the inspection	\$60	\$18,000

We estimate the following costs to accomplish any necessary adjustments that would be required based on the

results of the proposed inspections. We have no way of determining the number

of airplanes that may need such adjustments:

Labor cost	Parts cost	Total cost per airplane.
1 workhour × \$60 per hour	No parts necessary for adjustment	\$60

Regulatory Impact

Would this proposed AD impact various entities? The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposed rule would not have federalism implications under Executive Order 13132.

Would this proposed AD involve a significant rule or regulatory action? For the reasons discussed above, I certify that this proposed action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this

action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends Section 39.13 by adding a new airworthiness directive (AD) to read as follows:

British Aerospace:—Docket No. 99–CE–83–AD

- (a) What airplanes are affected by this AD? This AD affects Models HP137 Mk1, Jetstream series 200, and Jetstream Models 3101 and 3201 airplanes, all serial numbers, that are:
- (1) equipped with a nose landing gear unit 1873, B00A702852A, B00A703064A, or B00A703056A; and
 - (2) certificated in any category.
- (b) Who must comply with this AD? Anyone who wishes to operate any of the above airplanes on the U.S. Register must comply with this AD.
- (c) What problem does this AD address? The actions specified by this AD are intended to prevent cracked steering jack piston rods caused by inadequate clearance or inadequate lubrication of the steering jack pivot points. The condition could result in failure of the nose wheel steering system with consequent loss of airplane control.
- (d) What actions must I accomplish to address this problem? To address this problem, you must accomplish the following:

Action	Compliance time	Procedures
(1) Inspect the steering jack assembly to assure proper clearance between the bush heads on the steering plates and the shim on the steering jack trunnions and to assure that there is adequate lubrication at both trunnions and the eye end fitting.	Within the next 200 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.	Accomplish in accordance with the instructions in APPH Ltd. Service Newsletter, Issue 2, Jetstream 31 Steering Jack Part Number 618200, as referenced in British Aerospace the Mandatory Service Bulletin 32–JA 981043 dated March 5, 1999.
(2) Adjust the clearance and provide adequate lubrication, as necessary.	Prior to further flight after the inspection required by paragraph (d)(1) of this AD.	Accomplish in accordance with the instructions in APPH Ltd. Service Newsletter, Issue 2, Jetstream 31 Steering Jack Part Number 618200, as referenced in British Aerospace Mandatory Service Bulletin 32–JA 981043, dated March 5, 1999.

Note 1: The FAA is proposing other AD action that would require installing a modified nose landing gear steering actuator piston rod. Incorporating this action does not justify the need to inspect the piston rod for cracks.

- (e) Can I comply with this AD in any other way? You may use an alternative method of compliance or adjust the compliance time if:
- (1) Your alternative method of compliance provides an equivalent level of safety; and
- (2) The Manager, Small Airplane
 Directorate, approves your alternative.
 Submit your request through an FAA
 Principal Maintenance Inspector, who may
 add comments and then send it to the
 Manager, Small Airplane Directorate.
- **Note 2:** This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified,

altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

- (f) Where can I get information about any already-approved alternative methods of compliance? Contact Mr. S.M. Nagarajan, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4145; facsimile: (816) 329–4090.
- (g) What if I need to fly the airplane to another location to comply with this AD? The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.
- (h) How do I get copies of the documents referenced in this AD? You may obtain copies of the documents referenced in this AD from British Aerospace Regional Aircraft, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland. You may examine these documents at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106.

Note 3: The subject of this AD is addressed in British AD 012–03–99.

Issued in Kansas City, Missouri, on October 17, 2000.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00–27295 Filed 10–23–00; 8:45 am] **BILLING CODE 4910–13–P**

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-249-AD] RIN 2120-AA64

Airworthiness Directives; British Aerospace Model BAe ATP Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all British Aerospace Model BAe ATP airplanes. This proposal would require revising the Airworthiness Limitations Section of the Instructions for Continued Airworthiness to incorporate life limits for certain items and inspections to detect fatigue cracking in certain structures. This proposal is prompted by issuance of a revision to the airworthiness limitations of the British Aerospace ATP Aircraft Maintenance Manual, which specifies new inspections and compliance times for inspection and replacement action. The actions specified by the proposed AD are intended to ensure that fatigue cracking of certain structural elements is detected and corrected; such fatigue cracking could adversely affect the structural integrity of these airplanes.

DATES: Comments must be received by November 24, 2000.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-249-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9– anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 99-NM-249-AD" in the subject line and need not be submitted

in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from British Aerospace Regional Aircraft American Support, 13850 Mclearen Road, Herndon, Virginia 20171. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 99–NM–249–AD." The

postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 99–NM–249–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Discussion

The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom, has notified the FAA that a revision to Section 05-00-00 of British Aerospace ATP Aircraft Maintenance Manual (AMM) has been issued. [The FAA refers to the information included in that section of the AMM as the Airworthiness Limitations Section (ALS).] That revision affects all British Aerospace Model BAe ATP airplanes. The revision provides mandatory replacement times and structural inspection intervals approved under section 25.571 of the Joint Aviation Requirements and the Federal Aviation Regulations (14 CFR 25.571). As airplanes gain service experience, or as results of postcertification testing and evaluation are obtained, it may become necessary to add additional life limits or structural inspections to ensure the continued structural integrity of the airplane.

The CAA advises that analysis of fatigue test data has revealed that certain inspections must be performed at specific intervals to preclude fatigue cracking in certain areas of the airplane. In addition, the CAA advises that certain life limits must be imposed for various components on these airplanes to preclude the onset of fatigue cracking in those components. Such fatigue cracking, if not corrected, could adversely affect the structural integrity of these airplanes.

Explanation of Relevant Service Information

British Aerospace has issued a revision to Section 05-00-00, "General, Airworthiness Limitations," dated August 15, 1999, of British Aerospace ATP Aircraft Maintenance Manual (AMM), which references additional chapters. That revised section of the AMM includes mandatory life limitations for the airframe and power plant/engine; structural inspections of the fuselage, engine, horizontal stabilizer, and wing bottom surface. The revised section also describes new inspections and compliance times for inspection and replacement actions. Accomplishment of those actions will preclude the onset of fatigue cracking of