

Angeles Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

#### *Special Flight Permits*

(i) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### *Incorporation by Reference*

(j) The actions shall be done in accordance with McDonnell Douglas MD-80 Service Bulletin MD80-32-276, dated March 31, 1995; McDonnell Douglas MD-80 Service Bulletin MD80-32-276, Revision 1, dated October 17, 1995; McDonnell Douglas MD-80 Service Bulletin MD80-32-278, dated March 31, 1995; McDonnell Douglas MD-80 Service Bulletin MD80-32-278, Revision 1, dated September 6, 1995; McDonnell Douglas Service Bulletin DC9-32-289, dated March 7, 1996; McDonnell Douglas Service Bulletin DC9-32-311, dated July 6, 1998; or McDonnell Douglas Alert Service Bulletin DC9-32A311, Revision 1, dated March 8, 1999; as applicable.

(1) The incorporation by reference of McDonnell Douglas Service Bulletin DC9-32-311, dated July 6, 1998; or McDonnell Douglas Alert Service Bulletin DC9-32A311, Revision 1, dated March 8, 1999; is approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.

(2) The incorporation by reference of McDonnell Douglas MD-80 Service Bulletin MD80-32-276, dated March 31, 1995; McDonnell Douglas MD-80 Service Bulletin MD80-32-276, Revision 1, dated October 17, 1995; McDonnell Douglas MD-80 Service Bulletin MD80-32-278, dated March 31, 1995; and McDonnell Douglas MD-80 Service Bulletin MD80-32-278, Revision 1, dated September 6, 1995; was approved previously by the Director of the Federal Register as of February 26, 1996 (61 FR 2407, January 26, 1996).

(3) The incorporation by reference of McDonnell Douglas Service Bulletin DC9-32-289, dated March 7, 1996, was approved previously by the Director of the Federal Register as of November 14, 1996 (61 FR 53042, October 10, 1996).

(4) Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(k) This amendment becomes effective on November 12, 1999.

Issued in Renton, Washington, on September 28, 1999.

**D.L. Riffin,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 99-25766 Filed 10-6-99; 8:45 am]

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 98-NM-280-AD; Amendment 39-11351; AD 99-21-08]

RIN 2120-AA64

#### **Airworthiness Directives; Raytheon (Beech) Model 400A Airplanes**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain Raytheon (Beech) Model 400A airplanes, that requires replacement of the fuel drain tube assembly in the aft fuselage with a new, modified assembly. This amendment is prompted by a report of chafing of the fuel tube assembly against the elevator control cable due to inadequate clearance between the components. The actions specified by this AD are intended to prevent chafing of the fuel drain tube assembly, which could result in fuel leakage from the fuel drain tube assembly and consequent risk of a fire.

**DATES:** Effective November 12, 1999.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of November 12, 1999.

**ADDRESSES:** The service information referenced in this AD may be obtained from Raytheon Aircraft Company, Manager Service Engineering, Hawker Customer Support Department, P. O. Box 85, Wichita, Kansas 67201-0085. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Scott West, Aerospace Engineer,

Systems and Propulsion Branch, ACE-116W, FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946-4146; fax (316) 946-4407.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Raytheon (Beech) Model 400A airplanes was published in the **Federal Register** on August 10, 1999 (64 FR 43314). That action proposed to require replacement of the fuel drain tube assembly in the aft fuselage with a new, modified assembly.

#### **Comments**

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

#### **Conclusion**

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

#### **Cost Impact**

There are approximately 92 airplanes of the affected design in the worldwide fleet. The FAA estimates that 72 airplanes of U.S. registry will be affected by this AD, that it will take approximately 8 work hours per airplane to accomplish the required action, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$21 per airplane. Based on these figures, the cost impact of the required AD on U.S. operators is estimated to be \$36,072, or \$501 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

#### **Regulatory Impact**

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

#### 99-21-08 Raytheon Aircraft Company (Formerly Beech): Amendment 39-11351. Docket 98-NM-280-AD.

**Applicability:** Model 400A airplanes, serial numbers RK-1 through RK-92 inclusive, certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent chafing of the fuel drain tube assembly, which could result in fuel leakage from the fuel drain tube assembly and consequent risk of fire, accomplish the following:

#### Replacement

(a) At the next scheduled inspection, but no later than 200 flight hours after the effective date of this AD, replace the existing aft fuselage fuel drain tube assembly, part number (P/N) 128-920151-1, with a new, modified tube assembly, P/N 128-920237-1, in accordance with Raytheon Aircraft Service Bulletin SB.28-3076, dated October, 1997.

#### Spares

(b) As of the effective date of this AD, no person shall install a fuel drain tube assembly, P/N 128-920151-1, on any airplane.

#### Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA, Small Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

#### Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(e) The replacement shall be done in accordance with Raytheon Aircraft Service Bulletin SB.28-3076, dated October, 1997. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Raytheon Aircraft Company, Manager Service Engineering, Hawker Customer Support Department, P.O. Box 85, Wichita, Kansas 67201-0085. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(f) This amendment becomes effective on November 12, 1999.

Issued in Renton, Washington, on September 28, 1999.

**D.L. Riggins,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*  
[FR Doc. 99-25765 Filed 10-6-99; 8:45 am]  
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#### DEPARTMENT OF TRANSPORTATION

#### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 98-NM-267-AD; Amendment 39-11349; AD 99-21-06]

RIN 2120-AA64

#### Airworthiness Directives; McDonnell Douglas Model DC-9-81, -82, -83, and -87 Series Airplanes (MD-81, -82, -83, and -87), and Model MD-88 Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment supersedes an existing airworthiness directive (AD), applicable to all McDonnell Douglas Model DC-9-81, -82, -83, and -87 series airplanes (MD-81, -82, -83, and -87), and Model MD-88 airplanes, that currently requires visual or eddy current inspections to detect cracks of the actuator cylinder support brackets of the slat drive mechanism assembly, and replacement of any cracked brackets. This amendment continues to require repetitive eddy current inspection, adds an inspection requirement, and expands the area of inspection. This amendment also provides terminating action for the repetitive inspections. This amendment is prompted by reports indicating that additional cracking was found outside the original inspection area. The actions specified by this AD are intended to prevent inadvertent slat retraction in flight.

**DATES:** Effective November 12, 1999.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of November 12, 1999.

The incorporation by reference of McDonnell Douglas MD-80 Alert Service Bulletin A27-322, dated August 22, 1991, was approved previously by the Director of the Federal Register as of October 30, 1991 (56 FR 51645, October 15, 1991).

**ADDRESSES:** The service information referenced in this AD may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1-L51 (2-60). This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Transport Airplane Directorate, Los