other than the main office or a branch of the bank. A deposit production office (DPO) may solicit deposits, provide information about deposit products, and assist persons in completing application forms and related documents to open a deposit account. A DPO is not a branch within the meaning of 12 U.S.C. 36(j) and 12 CFR 5.30(d)(1) so long as it does not receive deposits, pay withdrawals, or make loans. All deposit and withdrawal transactions of a bank customer using a DPO must be performed by the customer, either in person at the main office or a branch office of the bank, or by mail, electronic transfer, or a similar method of transfer.

(b) Services of other persons. A national bank may use the services of, and compensate, persons not employed by the bank in its deposit production activities.

22. A new § 7.4005 is added to read as follows:

§ 7.4005 Combination of loan production office, deposit production office, and remote service unit.

A location at which a national bank operates a loan production office (LPO), a deposit production office (DPO), and a remote service unit (RSU) is not a "branch" within the meaning of 12 U.S.C. 36(j) by virtue of that combination. Since an LPO, DPO, or RSU is not, individually, a branch under 12 U.S.C. 36(j), any combination of these facilities at one location does not create a branch.

Dated: May 11, 1999.

John D. Hawke, Jr.,

Comptroller of the Currency.

[FR Doc. 99–14256 Filed 6–11–99; 8:45 am] BILLING CODE 4810–33–P

DEPARTMENT OF HOUSING AND URBAN DVELOPMENT

Office of Federal Housing Enterprise Oversight

12 CFR Part 1750 RIN 2550-AA02

Risk-Based Capital

AGENCY: Office of Federal Housing Enterprise Oversight, HUD.

ACTION: Proposed rule; extension of public comment period for the second notice of proposed rulemaking.

SUMMARY: On April 13, 1999, the Office of Federal Housing Enterprise Oversight (OFHEO) published a notice of proposed rulemaking entitled ''Risk-Based Capital'' in the **Federal Register** (64 FR 18083), the second of such

proposals related to the development of a regulation to establish risk-based capital standards for the Federal National Mortgage Association and the Federal Home Loan Mortgage Corporation. An earlier proposal, published on June 11, 1996, (61 FR 29592) set forth a methodology for identifying the benchmark credit loss experience specified by the Federal Housing Enterprise Financial Safety and Soundness Act of 1992 (1992 Act) and proposed the use of a House Price Index developed by OFHEO in the development of the stress test required by the 1992 Act. The second proposal (NPR 2) set forth the specifications for the stress test, completing OFHEO's risk-based capital proposal.

OFHEO has received several requests for an extension of the August 11, 1999, deadline for comments on NPR 2 to permit adequate time for interested parties to replicate and analyze the stress test and to understand the test as applied to a variety of possible starting points. In recognition of the complexity that necessarily attends this method of setting capital standards, the importance of a careful evaluation of the implications of this precedent-setting approach, and the value of meaningful comment in the rulemaking process, OFHEO is extending the comment period for NPR 2 from August 11, 1999, to November 10, 1999. This will insure that all interested parties have ample opportunity to participate in the rulemaking process by providing meaningful comment on the various technical and policy issues involved in the development of the risk-based capital regulation.

DATES: The comment period is extended until November 10, 1999.

ADDRESSES: Send written comments to Anne E. Dewey, General Counsel, Office of General Counsel, Office of Federal Housing Enterprise Oversight, 1700 G Street, NW., Fourth Floor, Washington, DC 20552. Written comments may also be sent by electronic mail to RegComments@OFHEO.gov.

FOR FURTHER INFORMATION CONTACT:

Patrick J. Lawler, Director of Policy Analysis and Chief Economist; David J. Pearl, Director, Research, Analysis and Capital Standards; or Gary L. Norton, Deputy General Counsel, Office of Federal Housing Enterprise Oversight, 1700 G Street, NW., Fourth Floor, Washington, DC 20552, telephone (202) 414–3800 (not a toll-free number). The telephone number for the Telecommunications Device for the Deaf is (800) 877–8339. Dated: June 9, 1999.

Mark A. Kinsey,

Acting Director, Office of Federal Housing Enterprise Oversight.

 $[FR\ Doc.\ 99{-}15002\ Filed\ 6{-}11{-}99;\ 8{:}45\ am]$

BILLING CODE 4220-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-CE-20-AD]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Models PC-12 and PC-12/ 45 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to certain Pilatus Aircraft Ltd. (Pilatus) Models PC-12 and PC-12/45 airplanes. The proposed AD would require replacing all flap drive shafts with flap drive shafts of improved design, installing additional gaskets on the power drive unit, and modifying the attachment and supporting hardware. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. The actions specified by the proposed AD are intended to prevent the flap drive shafts from corroding to the point where the flexible shafts in the flap drive system rupture, which could result in the inability to utilize the flap system with reduced airplane control.

DATES: Comments must be received on or before July 14, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 99–CE–20–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from Pilatus Aircraft Ltd., Customer Liaison Manager, CH–6371 Stans, Switzerland; telephone: +41 41 619 63 19; facsimile: +41 41 610 33 51. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. Roman T. Gabrys, Aerospace Engineer,

FAA, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone: (816) 426–6932; facsimile: (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 99–CE–20–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 99–CE–20–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Discussion

The Federal Office for Civil Aviation (FOCA), which is the airworthiness authority for Switzerland, recently notified the FAA that an unsafe condition may exist on certain Pilatus Models PC–12 and PC–12/45 airplanes. The FOCA of Switzerland reports several incidents of corroded inner drive cables of the flap flexible drive shafts.

Investigation of these incidents shows that moisture from the pressurized cabin can enter the flap drive shafts through the unsealed areas of the flap drive shaft attachments and the shaft casings. Over time, this moisture leads to corroded flap drive shafts.

This condition, if not corrected in a timely manner, could result in the flexible shafts in the flap drive system rupturing with possible reduced airplane control if the pilot lost the ability to utilize the flap system.

Relevant Service Information

Pilatus has issued Service Bulletin No. 27–003, dated March 8, 1999, which specifies procedures for replacing all flap drive shafts with flap drive shafts of improved design, installing additional gaskets on the power drive unit, and modifying the attachment and supporting hardware.

The FOČA of Switzerland classified this service bulletin as mandatory and issued Swiss AD HB 99–241, dated May 8, 1999, in order to assure the continued airworthiness of these airplanes in Switzerland.

The FAA's Determination

This airplane model is manufactured in Switzerland and is type certificated for operation in the United States under the provisions of § 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the FOCA of Switzerland has kept the FAA informed of the situation described above.

The FAA has examined the findings of the FOCA of Switzerland; reviewed all available information, including the service information referenced above; and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of the Provisions of the Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop in other Pilatus PC–12 and PC–12/45 airplanes of the same type design registered in the United States, the FAA is proposing AD action. The proposed AD would require replacing all flap drive shafts with flap drive shafts of improved design, installing additional gaskets on the power drive unit, and modifying the attachment and supporting hardware.

Accomplishment of the proposed action would be required in accordance with Pilatus Service Bulletin No. 27–003, dated March 8, 1999.

Compliance Time of the Proposed AD

The unsafe condition specified by the proposed AD is caused by corrosion. Corrosion can occur regardless of

whether the aircraft is being operated. For example, corrosion could develop on one of the affected airplanes at a certain time; then, if allowed to go undetected, the corrosion could develop into a more serious problem even if the airplane is in storage. Therefore, to assure that the unsafe condition specified in the proposed AD does not go undetected for a long period of time, the compliance is presented in calendar time instead of hours time-in-service (TIS).

Cost Impact

The FAA estimates that 69 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 19 workhours per airplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. Parts will be provided to the owners/operators of the affected aircraft free-of-charge. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$78,660, or \$1,140 per airplane.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation

Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

Pilatus Aircraft LTD.: Docket No. 99-CE-20-AD.

Applicability: Models PC-12 and PC-12/45 airplanes, manufacturer serial number (MSN) 101 through MSN 239, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated in the body of this AD, unless already accomplished.

To prevent the flap drive shafts from corroding to the point where the flexible shafts in the flap drive system rupture, which could result in the inability to utilize the flap system with reduced airplane control, accomplish the following:

(a) Within the next 4 calendar months after the effective date of this AD, accomplish the following in accordance with the Accomplishment Instructions section of Pilatus Service Bulletin No. 27–003, dated March 8, 1999:

- (1) Replace all flap drive shafts with flap drive shafts of improved design (part numbers as specified in paragraphs (b)(1) and (b)(2) of this AD);
- (2) Install additional gaskets on the power drive unit; and
- (3) Modify the attachment and supporting hardware.
- (b) As of the effective date of this AD, no person may install, on any affected airplane, a flap drive shaft assembly that is not of the following part numbers (or FAA-approved equivalent part numbers):
- (1) Part number 945.02.02.201: Flap Drive Shaft 953D100–5 (Inboard); and
- (2) Part-number 945.02.02.202: Flap Drive Shaft 953D100–7 (Outboard).

Note 2: The FAA recommends that the owner/operator of the affected airplanes insert Pilatus Temporary Revision No.27–07, dated January 8, 1999, into the PC12

Maintenance Manual at the same time this AD is accomplished to assure that the maintenance procedures for the improved design parts are current.

(c) Special flight permits may be issued in accordance with $\S\S21.197$ and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance times that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(e) Questions or technical information related to Pilatus Service Bulletin No: 27–003, dated March 8, 1999, should be directed to Pilatus Aircraft Ltd., Customer Liaison Manager, CH–6371 Stans, Switzerland; telephone: +41 41 619 63 19; facsimile: +41 41 610 33 51. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Note 4: The subject of this AD is addressed in Swiss AD HB 99–241, dated May 8, 1999.

Issued in Kansas City, Missouri, on June 4, 1999.

Marvin R. Nuss,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 99–14937 Filed 6–11–99; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-CE-10-AD]

RIN 2120-AA64

Airworthiness Directives; Pilatus Aircraft Ltd. Models PC-12 and PC-12/ 45 Airplanes

AGENCY: Federal Aviation Administration. DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to certain Pilatus Aircraft Ltd. (Pilatus) Models PC–12 and PC–12/45 airplanes. The proposed AD would require modifying the generator 2 excitation by removing certain diodes

and installing a new 5-amp circuit breaker and suppression filter. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Switzerland. The actions specified by the proposed AD are intended to prevent damage to electrical components if the generator 2 is not switched off prior to engine shutdown and it overheats, which could result in loss of electrical power to certain critical airplane components.

DATES: Comments must be received on or before July 14, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 99–CE–10–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from Pilatus Aircraft Ltd., Customer Liaison Manager, CH–6371 Stans, Switzerland; telephone: +41 41 619 63 19; facsimile: +41 41 610 33 51. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. Roman T. Gabrys, Aerospace Engineer, FAA, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone: (816) 426–6932; facsimile: (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this