provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch,

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(f) The actions shall be done in accordance with Airbus A319/320/321 Flight Manual Temporary Revision (TR) 2.05.00/13, dated October 18, 1996; and Airbus All Operators Telex (AOT) 34-03, dated February 20, 1996. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington,

Note 3: The subject of this AD is addressed in French airworthiness directive (CN) 96-172-084(B), dated August 28, 1996.

(g) This amendment becomes effective on January 20, 1998.

Issued in Renton, Washington, on December 5, 1997.

#### John J. Hickey.

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 97-32430 Filed 12-12-97; 8:45 am] BILLING CODE 4910-13-U

## **DEPARTMENT OF TRANSPORTATION**

## **Federal Aviation Administration**

## 14 CFR Part 39

[Docket No. 97-NM-146-AD: Amendment 39-10241; AD 97-25-16]

RIN 2120-AA64

## Airworthiness Directives; British Aerospace (Jetstream) Model 4101 **Airplanes**

AGENCY: Federal Aviation Administration, DOT. **ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain British Aerospace (Jetstream) Model 4101 airplanes, that

requires a one-time inspection of the tailplane (horizontal stabilizer) leading edges and surrounding area for discrepancies, and corrective action, if necessary. This amendment is prompted by the issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent separation of the horizontal stabilizer from the fin, which could lead to reduced controllability of the airplane.

DATES: Effective January 20, 1998. The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of January 20, 1998.

**ADDRESSES:** The service information referenced in this AD may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: International Branch, ANM-116, FAA. Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain British Aerospace (Jetstream) Model 4101 airplanes was published in the Federal Register on October 1, 1997 (62 FR 51386). That action proposed to require a one-time inspection of the tailplane (horizontal stabilizer) leading edges and surrounding area for discrepancies, and corrective action, if necessary.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

#### Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

#### **Cost Impact**

The FAA estimates that 57 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1

work hour per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$3,420, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

## **Regulatory Impact**

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034. February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

## **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### **Adoption of the Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## **PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

#### 97-25-16 British Aerospace Regional Aircraft [Formerly Jetstream Aircraft Limited, British Aerospace (Commercial Aircraft) Limited]: Amendment 39-10241. Docket 97-NM-146-AD.

Applicability: Jetstream Model 4101 airplanes, constructors numbers 41005 through 41069 inclusive, and 41071 through 41078 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent separation of the tailplane (horizontal stabilizer) from the fin, which could lead to reduced controllability of the airplane, accomplish the following:

- (a) Within 3,000 hours time-in-service or 11 months after the effective date of this AD, whichever occurs first, perform a one-time inspection of the attachment bolts and surrounding area of the horizontal stabilizer leading edges for discrepancies, in accordance with Jetstream Service Bulletin J41–55–002, Revision 1, dated July 25, 1996. If any discrepancy is found, prior to further flight, correct the discrepancy in accordance with the service bulletin.
- (b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116

- (c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (d) The actions shall be done in accordance with Jetstream Service Bulletin J41–55–002, Revision 1, dated July 25, 1996, which contains the specified effective pages:

Page No.	Revision level shown on page	Date shown on page
1, 3	1	July 25, 1996.
2, 4–9	Original	June 28, 1996.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed British airworthiness directive 004–06–96.

(e) This amendment becomes effective on January 20, 1998.

Issued in Renton, Washington, on December 5, 1997.

#### John J. Hickey,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 97–32431 Filed 12–12–97; 8:45 am] BILLING CODE 4910–13–U

## **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

## 14 CFR Part 39

[Docket No. 97-NM-161-AD; Amendment 39-10243; AD 97-25-18]

## RIN 2120-AA64

# Airworthiness Directives; Aerospatiale Model ATR42 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Aerospatiale Model ATR42 series airplanes, that requires removal of certain landing gear attachment pins, and replacement of the pins with serviceable pins. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent wear of the attachment pins, which could result in collapse of the main landing gear.

DATES: Effective January 20, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of January 20, 1998.

ADDRESSES: The service information referenced in this AD may be obtained from Aerospatiale, 316 Route de Bayonne, 31060 Toulouse, Cedex 03, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

## FOR FURTHER INFORMATION CONTACT: International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Aerospatiale Model ATR42 series airplanes was published in the **Federal Register** on August 25, 1997 (62 FR 44915). That action proposed to require removal of certain landing gear attachment pins, and replacement of the pins with serviceable pins.

#### **Comments**

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter supports the proposed rule.

#### **Conclusion**

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

## Cost Impact

The FAA estimates that 88 Model ATR42 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 45 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will be provided by the manufacturer at no cost to operators. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$237,600, or \$2,700 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.